

Steel
رابعه انشادات
ماروزى



٥٣

Final Term Revision

Part 3

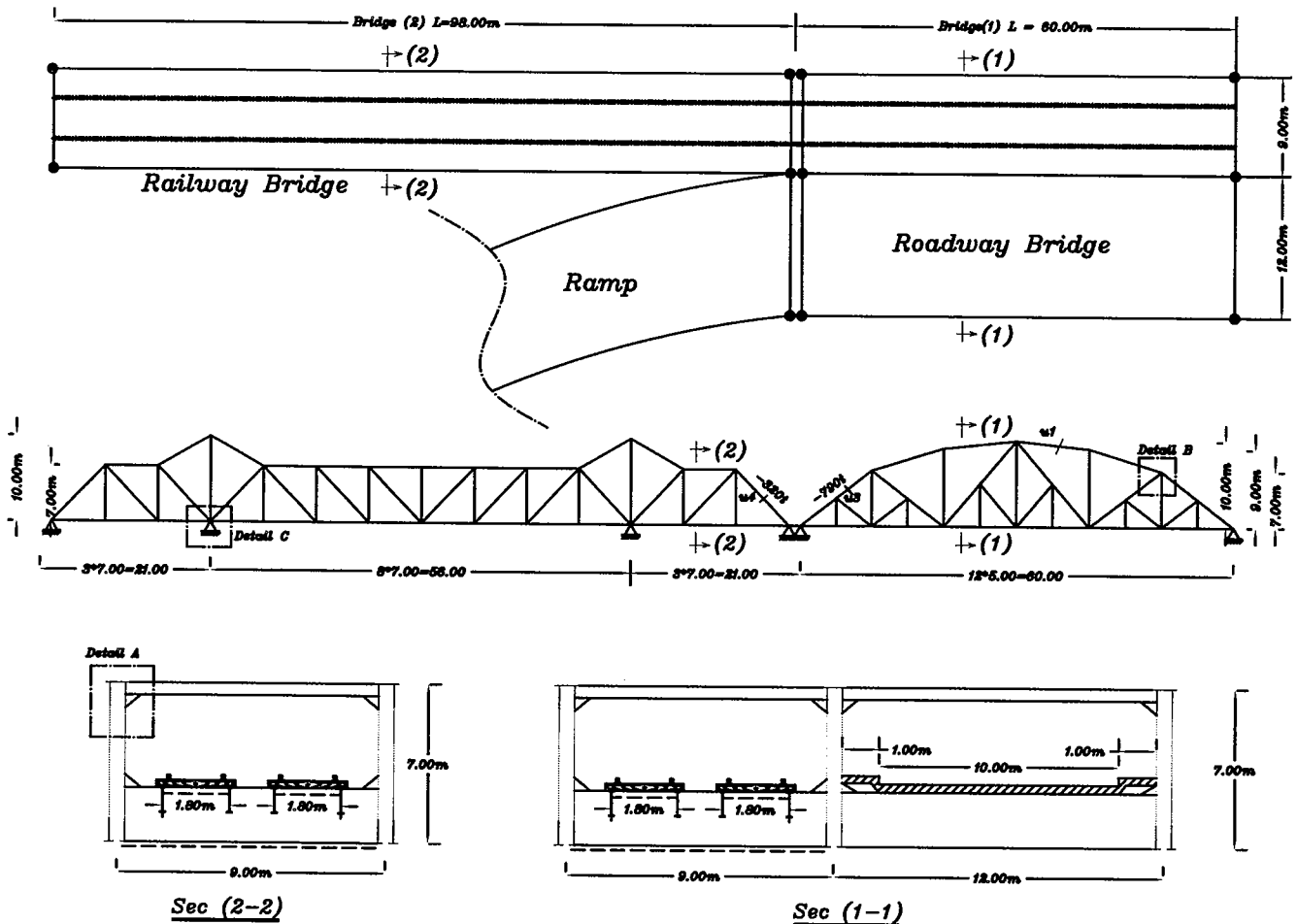
Examples

2013-2014

Example one

Design of steel Bridges

- Material of Construction is steel 44
- any missing data may be reasonably assumed



A combined road-railway through bridge is adjoined at Bridge (1), with three main camel-back truss system, as shown in the figure, Railway is double-track line and the roadway is 10.00m wide with two sidewalks 1.00m each as shown in section (1-1). Only the rail way line is continual over Bridge 2 with the through truss configuration as shown in the figure above it is required to :

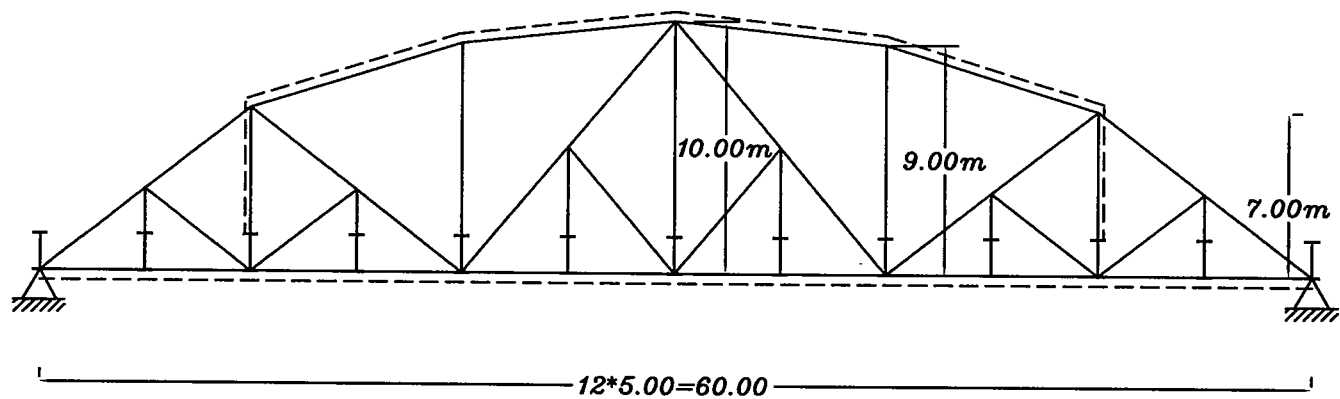
- 1-Draw a complete general layout for both bridges (1)&(2) to reasonable scale (plans, elevations & cross sections) showing the required bracings of both bridges

2-Draw to scale 1:10 .assume any missind Data (DetailA,B,C)

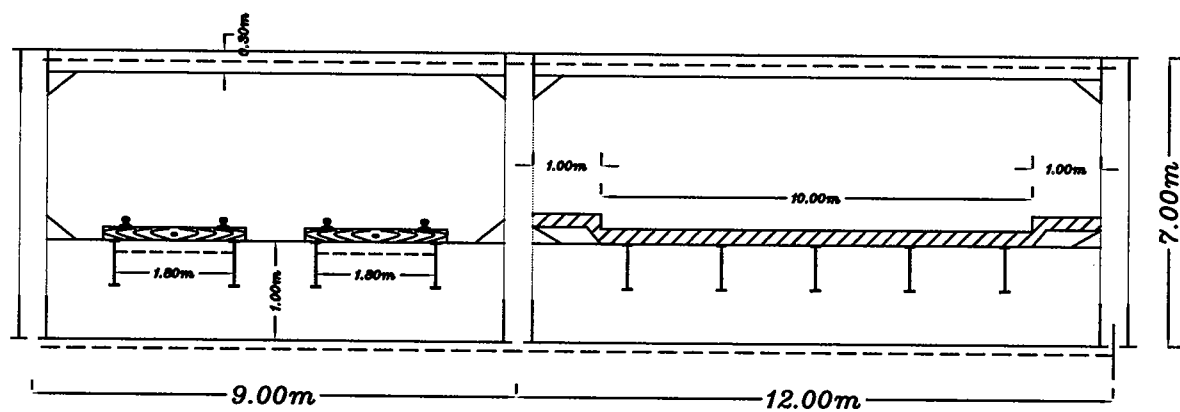
3-For the marked member calculate the max. forces due to
Live load+Impact

4-Design the two marked members U3 , U4.

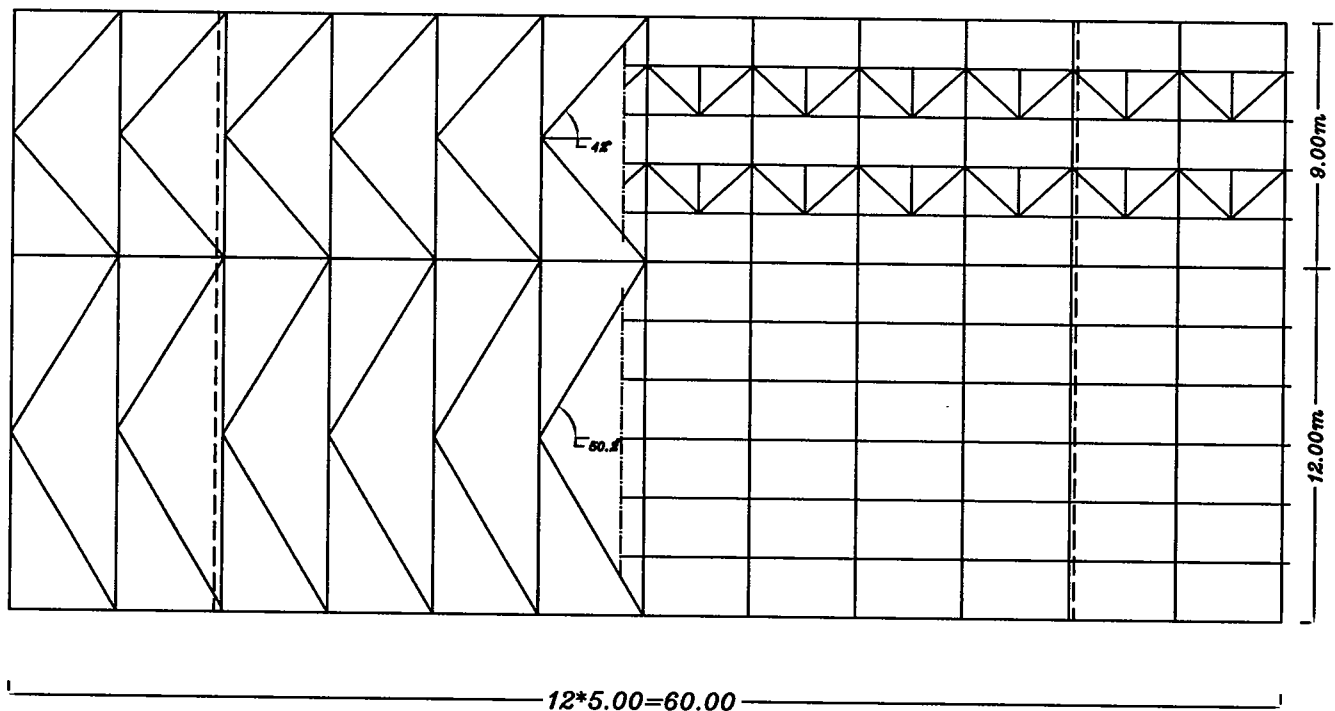
Bridge one



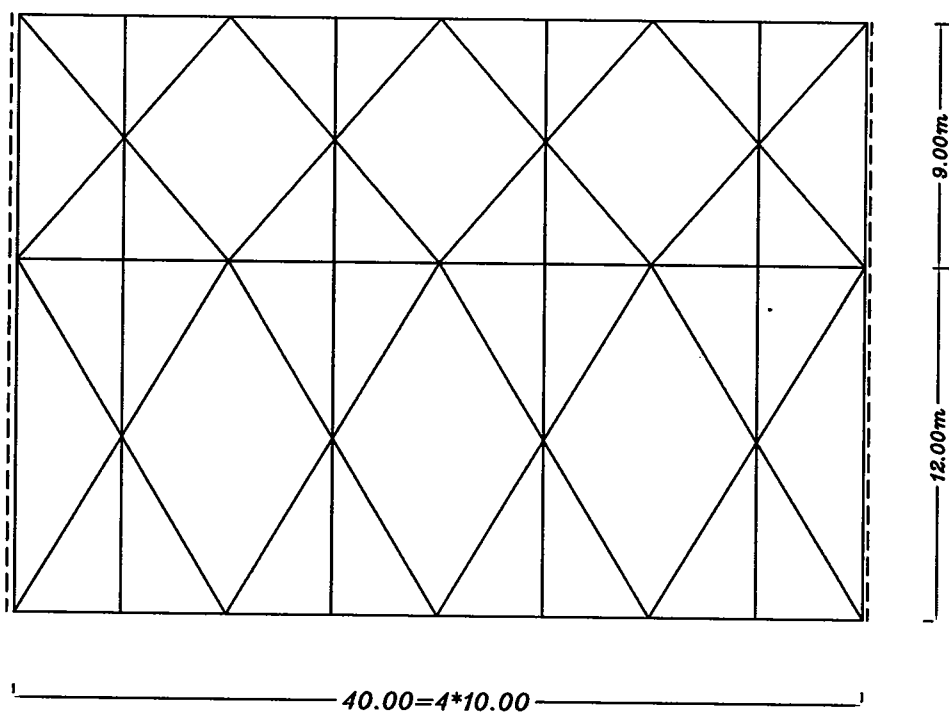
Elevation



Sec (1-1)

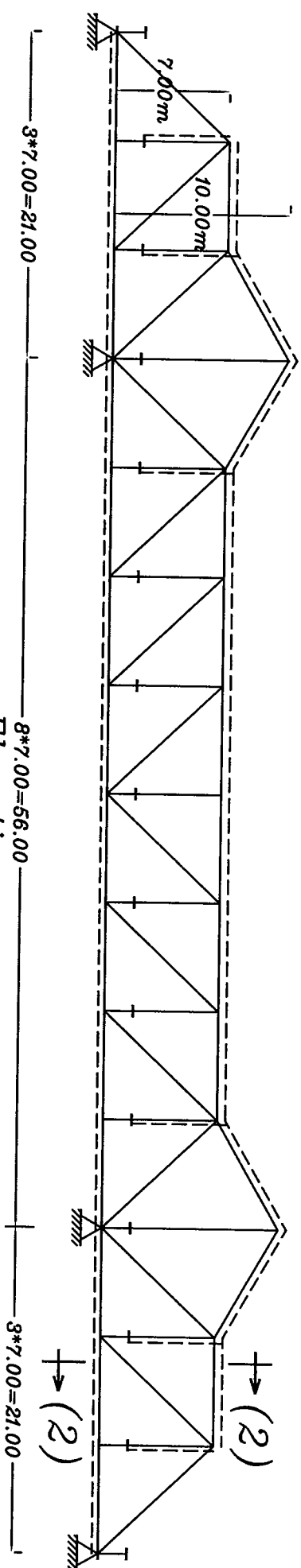


Plan of lower bracing

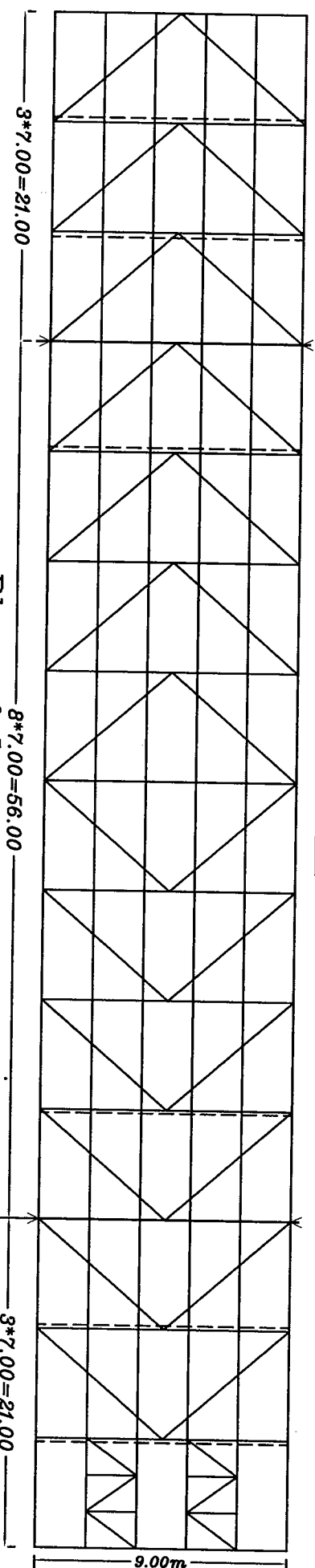


Plan of upper bracing

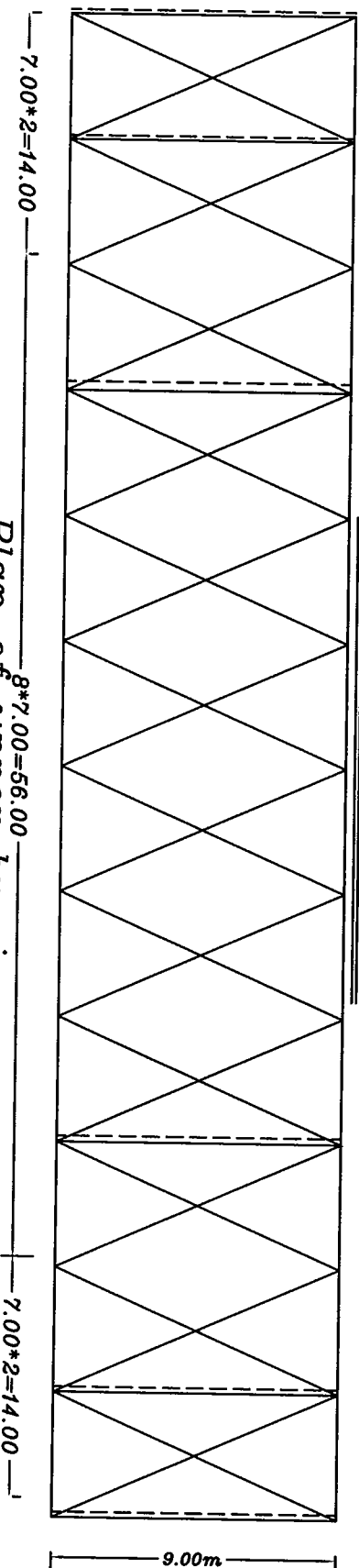
Bridge two



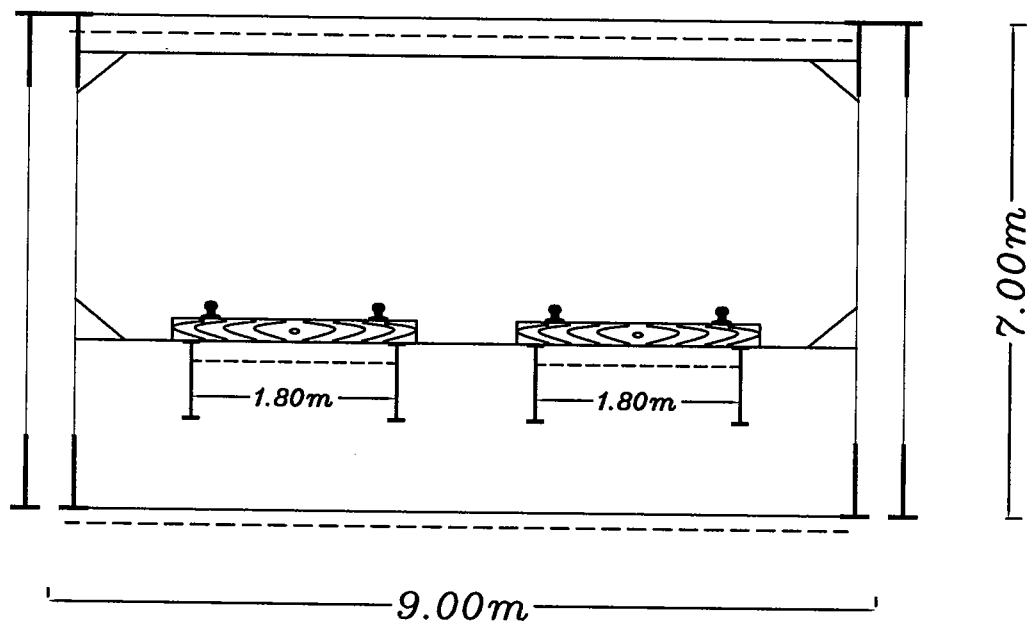
Elevation



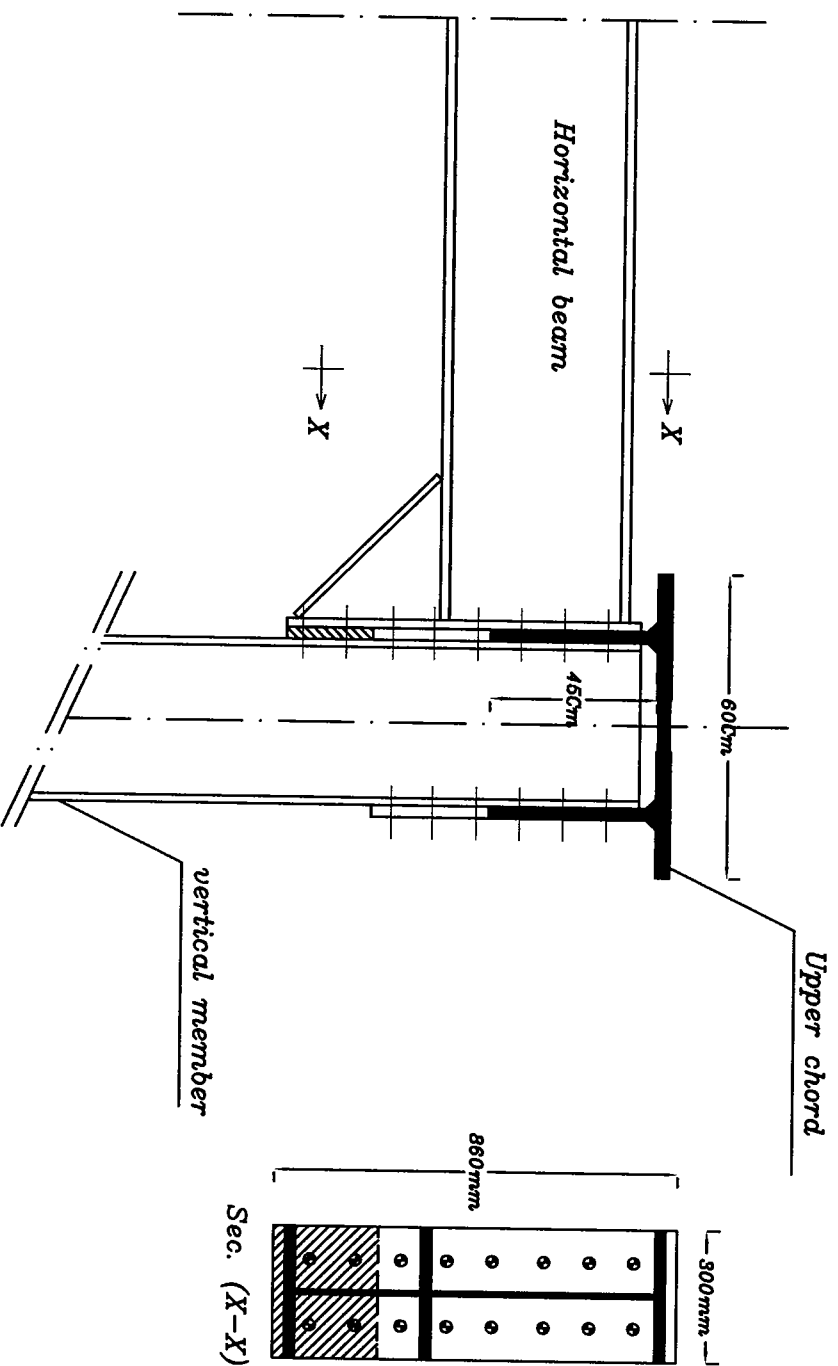
Plan of Lower bracing



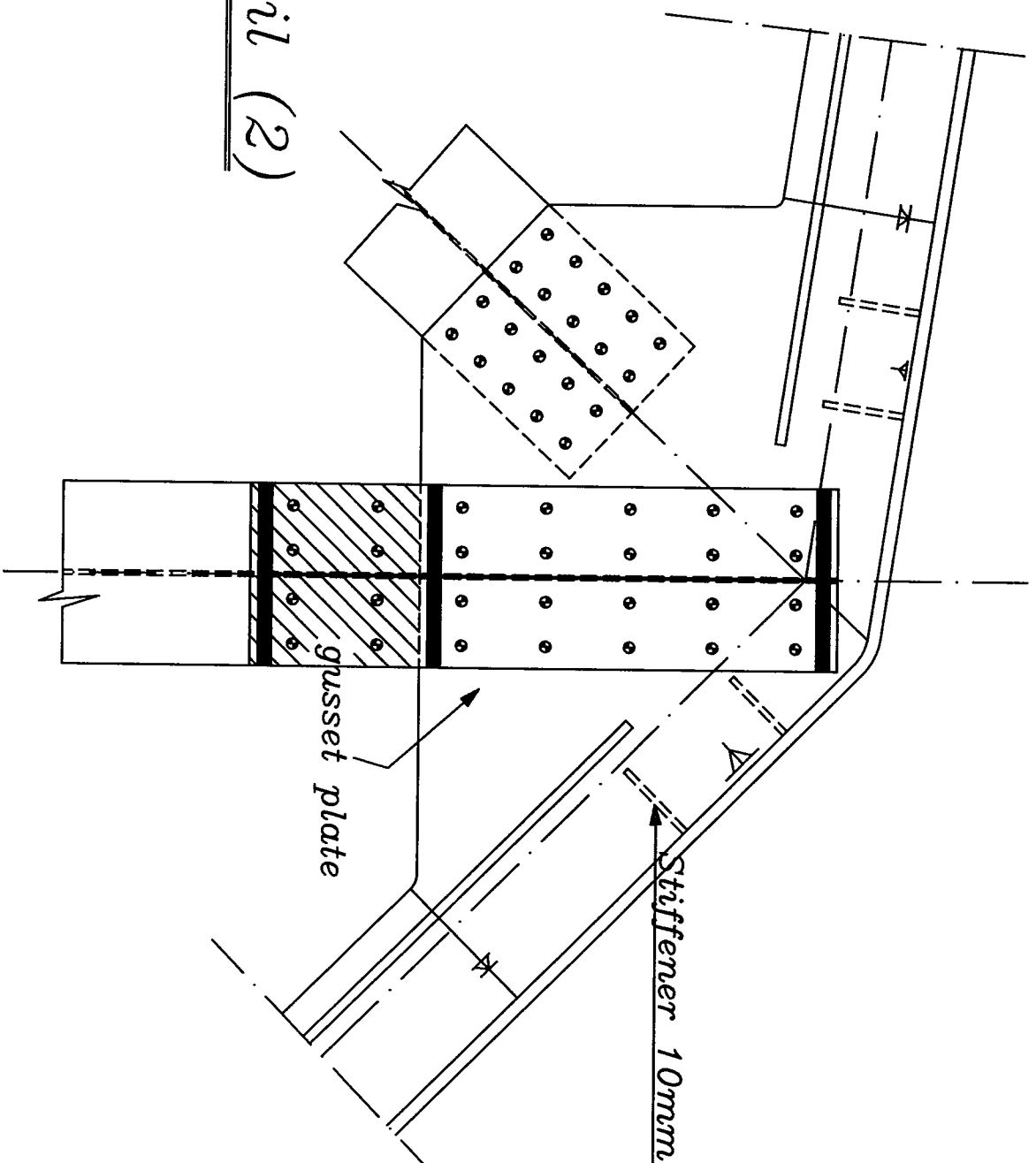
Plan of upper bracing



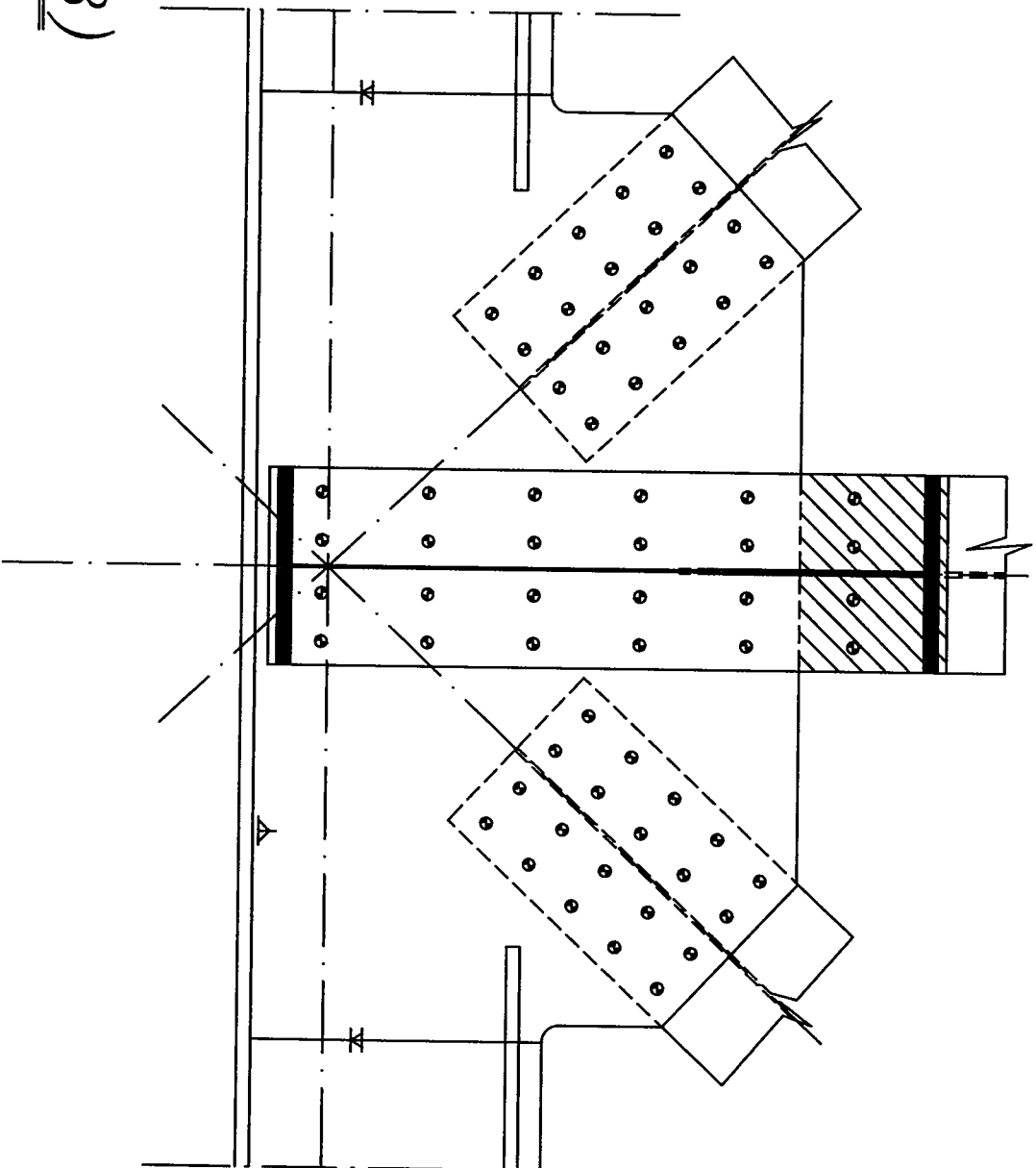
Sec (2-2)



Detail (1)

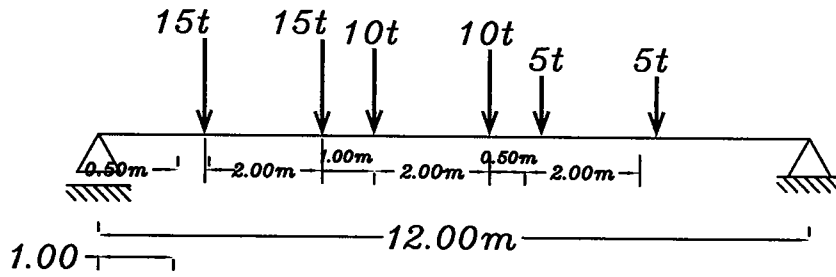


Detail (3)

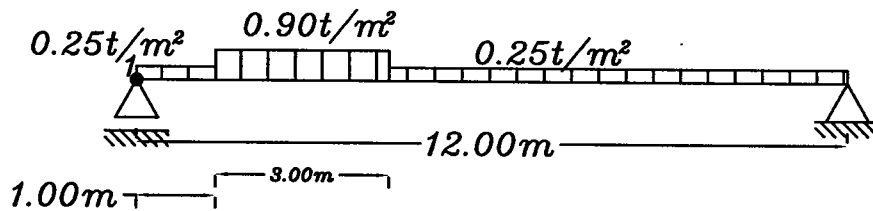


Question 3

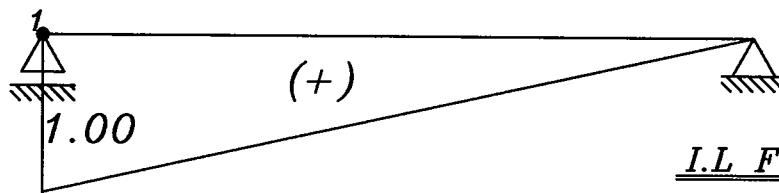
For road way part



Strip1
Get R



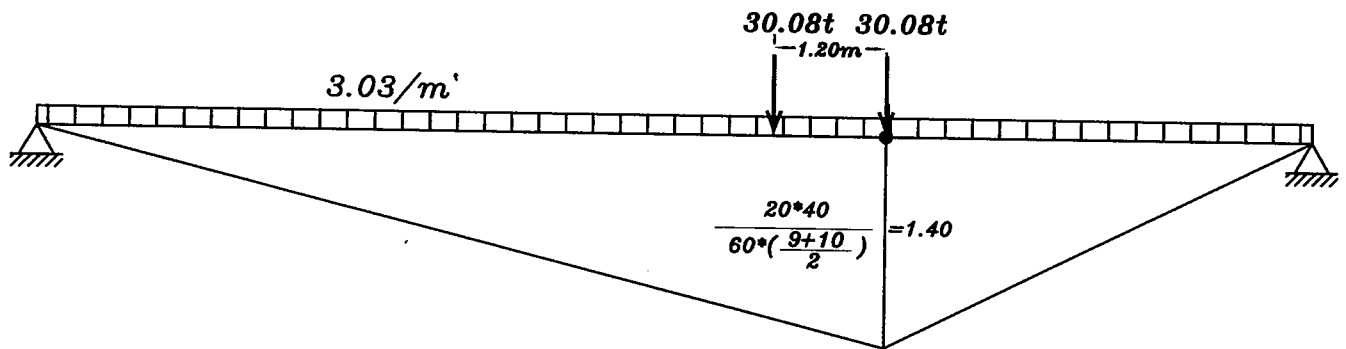
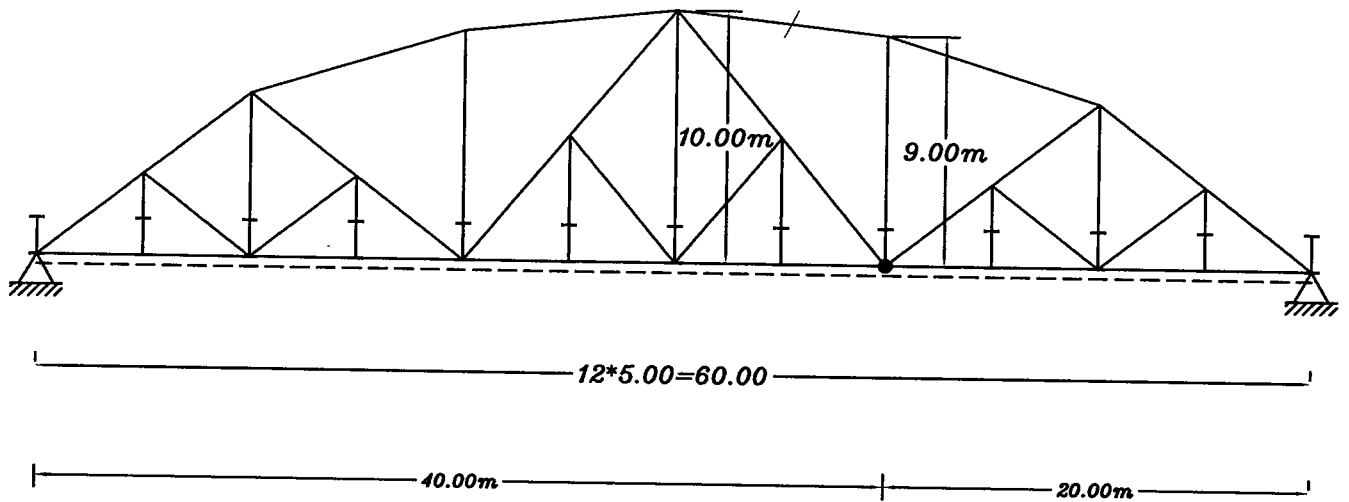
Strip2
Get W



I.L For Reaction
@ 1

$$R_1 = 15 * \left(\frac{10.5 + 8.5}{12} \right) + 10 * \left(\frac{7.5 + 5.5}{12} \right) + 5 * \left(\frac{4.5 + 2.5}{12} \right) = \boxed{30.08t}$$

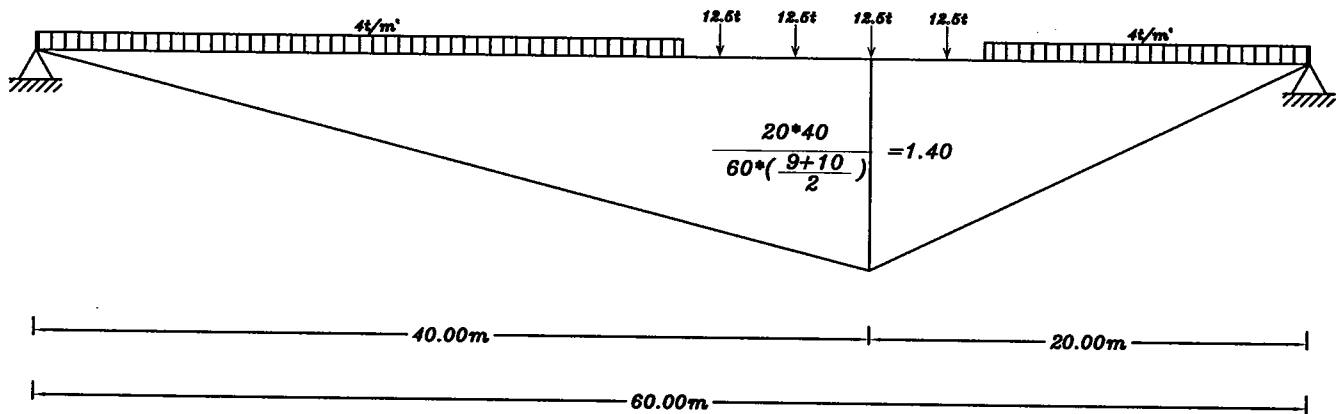
$$W_1 = 0.25 * 1.0 * 0.958 + 0.9 * 3 * 0.79 + 0.25 * 8.00 * 0.33 = \boxed{3.032t/m^2}$$



$$F_{LL+I} = 30.08 \times 1.40 + 30.08 \times 1.358 + 3.03 \times 0.5 \times 1.40 \times 60 = \boxed{210.22t}$$

Road

For rail way part



$$F_{LL, \text{Rail}} = 3 \times 12.5 \times 1.344 + 12.5 \times 1.288 + 4(36 \times 0.63 + 17.6 \times 0.616) = \boxed{200.58t}$$

$$(1+I) = 0.73 + \frac{2.16}{\sqrt{60.0} - 0.2} = 1.01 \quad \boxed{< 1.1, \text{ use } 1.1}$$

$$F_{LL+I, \text{Rail}} = 200.58 \times 2 \times 1.1 = \boxed{441.276t}$$

$$F_{LL+I, \text{Rail+Road}} = 441.276t + 210.22t = \boxed{-651.496t \text{ Comp.}}$$

Question 4

Design of U3 (-790t Comp.)

1) assume compression stress to be :

$$F_c = 1.3t/\text{Cm}^2 \text{ For St.44}$$

$$h = \frac{\text{Pannel length}}{10 \rightarrow 15} = \frac{500}{10 \rightarrow 15}$$

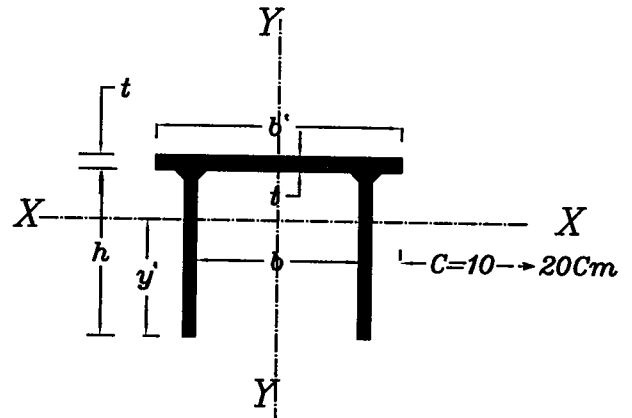
$$h = 50\text{Cm to } 33.3\text{Cm}$$

$$\text{use } h = 45\text{Cm}$$

$$\text{use } b = 45\text{Cm}$$

$$b' = b + 2*(10 \rightarrow 20\text{Cm})$$

$$b' = 45 + 2*(10 \rightarrow 20\text{Cm}) = 65\text{Cm to } 85\text{Cm use } b' = 65\text{Cm}$$



$$2) F_c = \frac{\text{max. force}}{\text{area}} \quad 1.3 = \frac{790}{\text{area}}$$

$$\text{area} = 607.7\text{Cm}^2$$

$$A = 2ht + b'*t = \dots \text{Cm}^2$$

$$607.7 = 2*45*t + 65*t, \quad t = 3.92\text{Cm}$$

$$\text{take } t = 4.00\text{Cm}$$

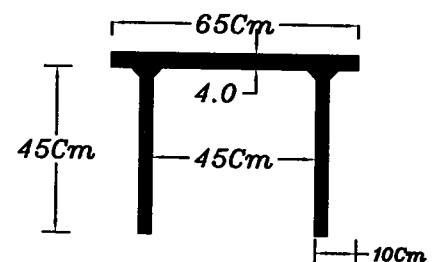
Checks

1-Check Compactness

$$\frac{b}{t} = \frac{45 - 2*1}{4.0} = 10 < \frac{64}{\sqrt{2.8}} = 38.2$$

$$\frac{C}{t} = \frac{10 - 4.0 - 1}{4.0} = 1.2 < \frac{21}{\sqrt{2.8}} = 12.5$$

$$\frac{h}{t} = \frac{45 - 1}{4.0} = 11 < \frac{30}{\sqrt{2.8}} = 17.9$$



2-Check global buckling

$$y' = \frac{(2ht*0.5h)+b'*t(h+0.5t)}{2ht+b't} = \dots\dots Cm$$

$$y' = \frac{(45*4.0*22.5*2)+(65*4*47.5)}{(45*4.0*2)+(65*4)} = 32.9Cm$$

$$I_x = 2 * \frac{t*h^3}{12} + 2*t*h*(y'-0.5h)^2 + b'*t*(y'-h-0.5t)^2 = \dots\dots Cm^4$$

$$I_x = 2 * \frac{4.0*45^3}{12} + 2*4.0*45*(32.9-22.5)^2 + 4.0*65*(47-32.9)^2$$

$$I_x = 151378 \quad Cm^4$$

$$I_y = \left[\frac{4.0*65^3}{12} \right] + 2*4.0*45*(22.5+2)^2 = 307631 \quad Cm^4$$

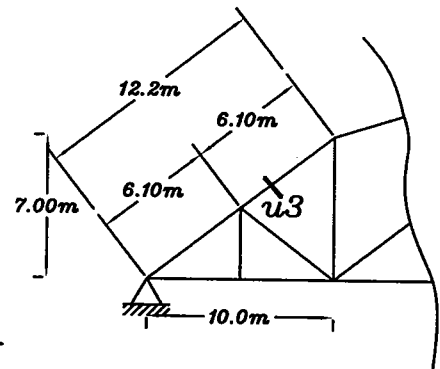
$$A = (4.0*2*45) + (65*4.0) = 620 \quad Cm^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{151378}{620}} = 15.62Cm$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{307631}{620}} = 22.27Cm$$

$$\lambda_{in} = \frac{0.85L}{r_x} = \frac{0.85*610}{15.62} = 33.19 < 90$$

$$\lambda_{out} = \frac{L_{out}}{r_y} = \frac{0.85*1220}{22.27} = 46.56 < 90$$



3-Check Compressive Stresses

$$\text{actual stresses} = f_{ca} = \frac{790}{620} = 1.27 \quad t/Cm^2$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * \lambda_{max}^2 \quad \text{For St.44}$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * 46.5^2 = 1.41 \quad t/Cm^2$$

Question 4

Design of U4 (-320t Comp.)

1) assume compression stress to be :

$$F_c = 1.3t/Cm^2 \text{ For St.44}$$

$$h = \frac{\text{Pannel length}}{10 \rightarrow 15} = \frac{700}{10 \rightarrow 15}$$

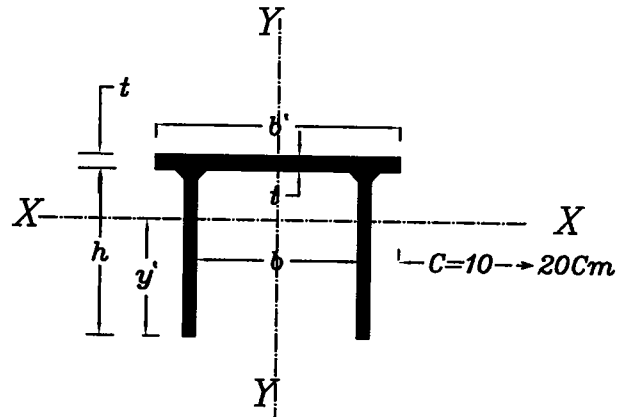
$$h = 70Cm \text{ to } 46.6Cm$$

$$\text{use } h = 55Cm$$

$$\text{use } b = 55Cm$$

$$b' = b + 2*(10 \rightarrow 20Cm)$$

$$b' = 55 + 2*(10 \rightarrow 20Cm) = 75Cm \text{ to } 95Cm \text{ use } b' = 75Cm$$



$$2) F_c = \frac{\text{max. force}}{\text{area}} \quad 1.3 = \frac{320}{\text{area}}$$

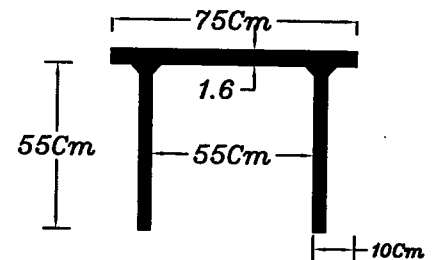
$$\text{area} = 246.1Cm^2$$

$$A = 2ht + b'*t = \dots Cm^2$$

$$246.1 = 2*55*t + 55*t$$

$$\text{take } t = 1.60Cm$$

$$, t = 1.49Cm \quad \text{use } t = 16mm$$



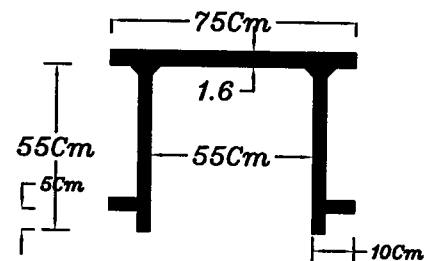
Checks

1-Check Compactness

$$\frac{b}{t} = \frac{55 - 2*1}{1.6} = 33 < \frac{64}{\sqrt{2.8}} = 38.2$$

$$\frac{C}{t} = \frac{10 - 1.6 - 1}{1.6} = 4.6 < \frac{21}{\sqrt{2.8}} = 12.5$$

$$\frac{h}{t} = \frac{55 - 1}{1.6} = 33 > \frac{30}{\sqrt{2.8}} = 17.9 \text{ unsafe use stiffener}$$



2-Check global buckling

$$y' = \frac{(2ht*0.5h)+b'*t(h+0.5t)}{2ht+b't} = \dots\dots Cm$$

$$y' = \frac{(55*1.6*27.5*2)+(75*1.6*55.8)}{(55*1.6*2)+(75*1.6)} = 38.9 Cm$$

$$I_x = 2 * \frac{t*h^3}{12} + 2*t*h*(y'-0.5h)^2 + b'*t*(y'-h-0.5t)^2 = \dots\dots Cm^4$$

$$I_x = 2 * \frac{1.6*55^3}{12} + 2*1.6*55*(38.9-27.5)^2 + 1.6*75*(55.8-38.9)^2$$

$$I_x = 101512 \quad Cm^4$$

$$I_y = \left[\frac{1.6*75^3}{12} \right] + 2*1.6*55*(27.5+0.8)^2 = 197206 \quad Cm^4$$

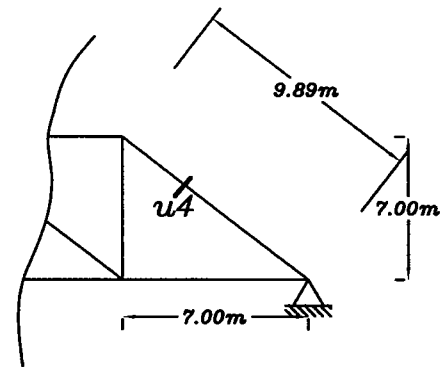
$$A = (1.6*2*55) + (75*1.6) = 296 \quad Cm^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{101512}{296}} = 18.51 Cm$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{197206}{296}} = 25.81 Cm$$

$$\lambda_{in} = \frac{0.85L}{r_x} = \frac{0.85*989}{15.62} = 53.81 < 90$$

$$\lambda_{out} = \frac{L_{out}}{r_y} = \frac{0.85*989}{22.27} = 37 < 90$$



3-Check Compressive Stresses

$$\text{actual stresses} = f_{ca} = \frac{320}{296} = 1.08 \quad t/Cm^2$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * \lambda_{max}^2 \quad \text{For St.44}$$

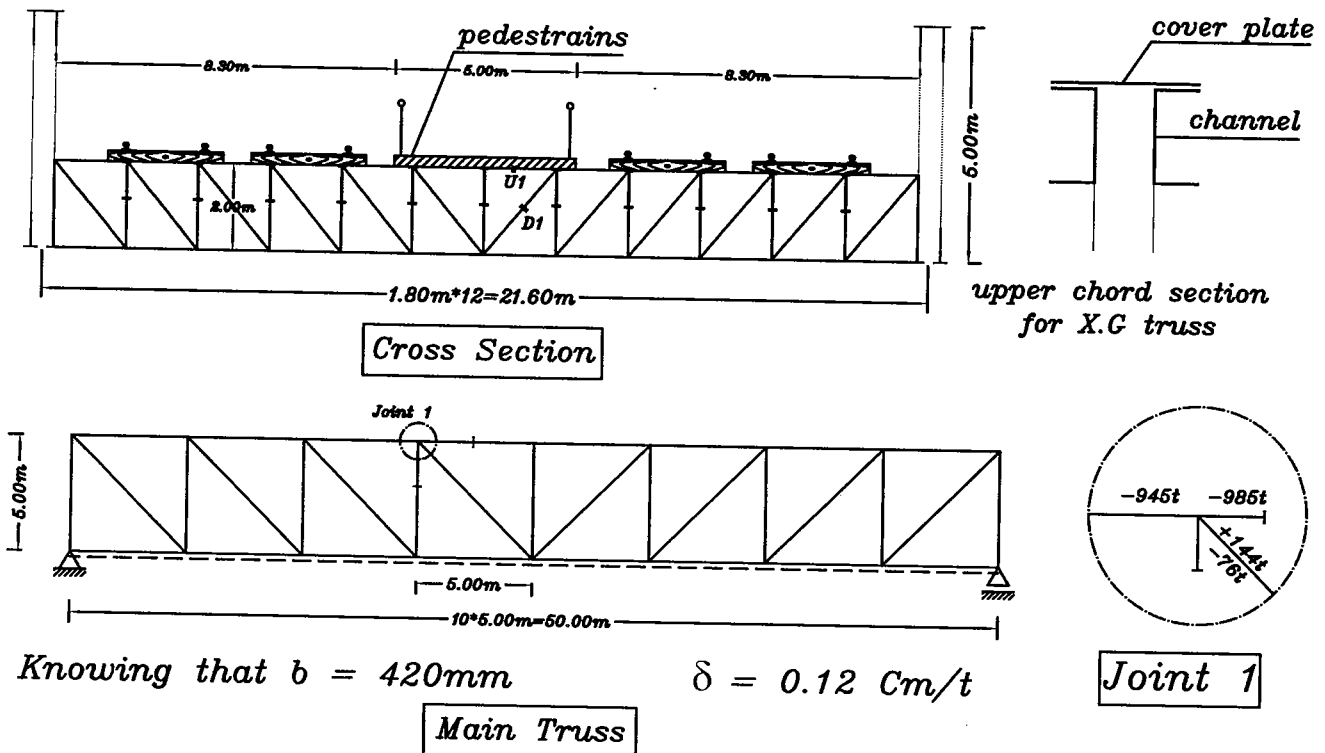
$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * 53.8^2 = 1.35 \quad t/Cm^2$$

Example two

Design of steel Bridges

- Material of Construction is steel 44
- any missing data may be reasonably assumed

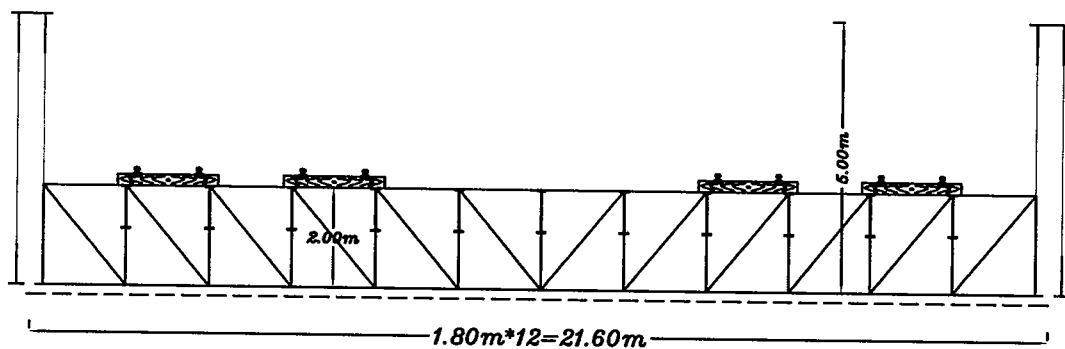
For the shown Road-rail way pony bridge having a X.G – truss of a 21.60m span and span of the bridge is 50.00m, spacing between X.G is 5.00m it is required to:



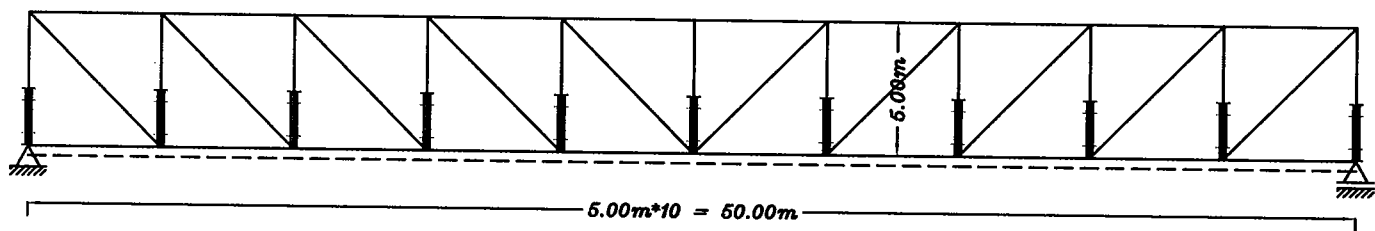
- 1-for the above mentioned bridge draw to scale 1:100 Complete General lay out.
- 2-Calculate the design forces for the upper chord U1 (x.g).
- 3-Calculate max. and min. force due to LL+I Only for D1 (x.g).
- 4-design a suitable double channel section for the upper chord.
- 5-design a suitable section for two marked member in the main truss.

the live load on roadway part from pedestains and it equal 300Kg/m^2

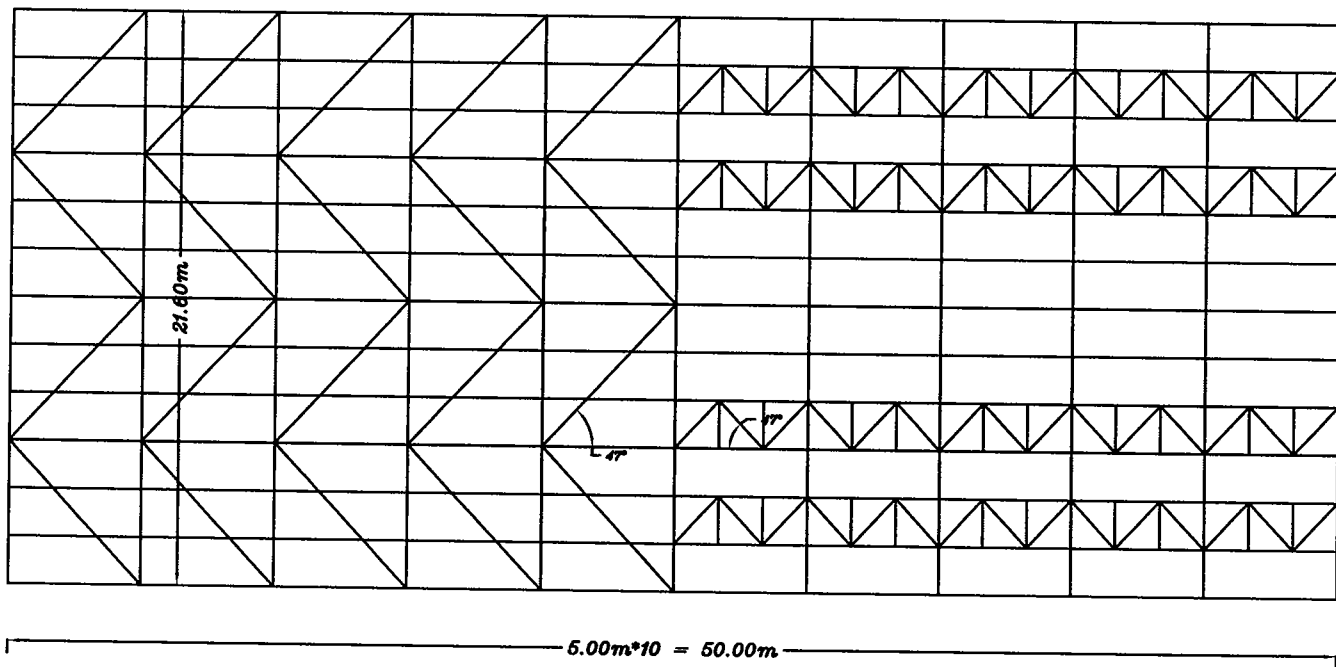
Question 1



Cross Section



Elevation



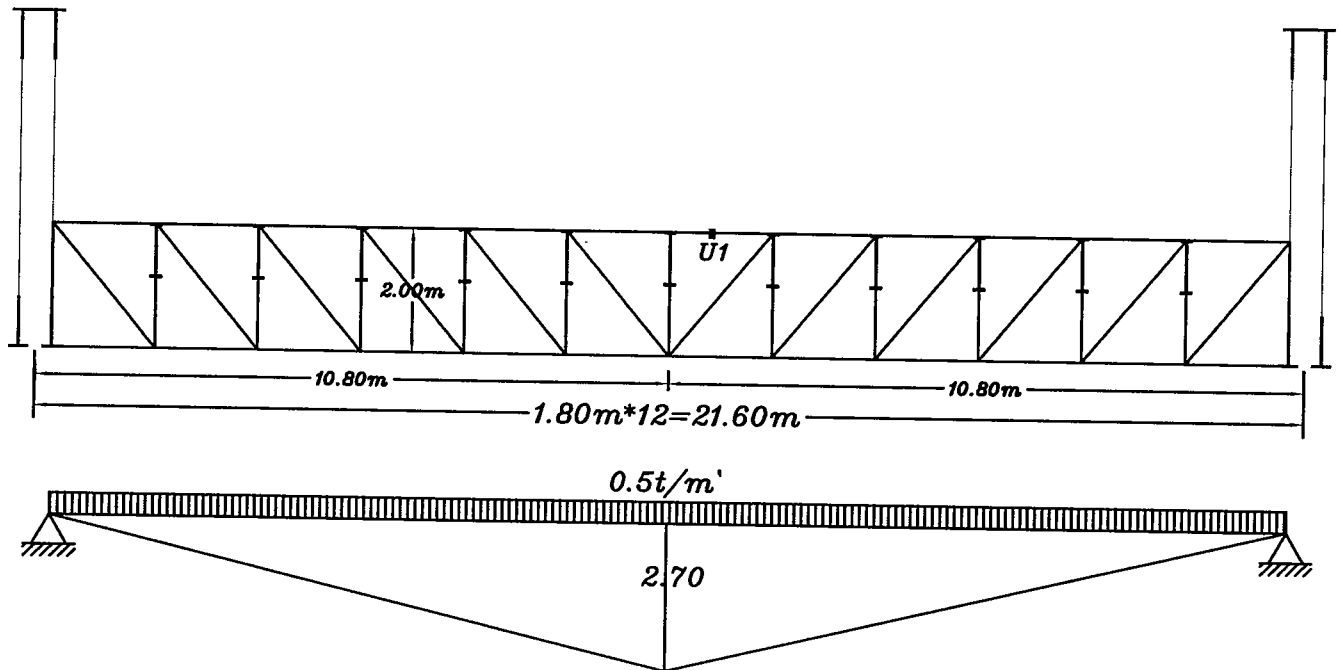
Plan of Lower Bracing

Question 2

upper chord u1

Dead loads

1-assume o.w of the truss is $0.5t/m'$



$$F_{o.w} = 0.5 \times 21.60 \times 2.70 \times 0.5 = 14.58t$$

2-due to rail and road part

2-i rail way part

$$W_{Dead} = 600/2 + 40/2 + 0.W = \dots\dots Kg/m'$$

$$W_{Dead} = 0.6/2 + 0.04/2 + 0.15 = 0.47t/m'$$

$$R_d = W_d * S = 0.47 * 5.00 = \boxed{2.35t}$$

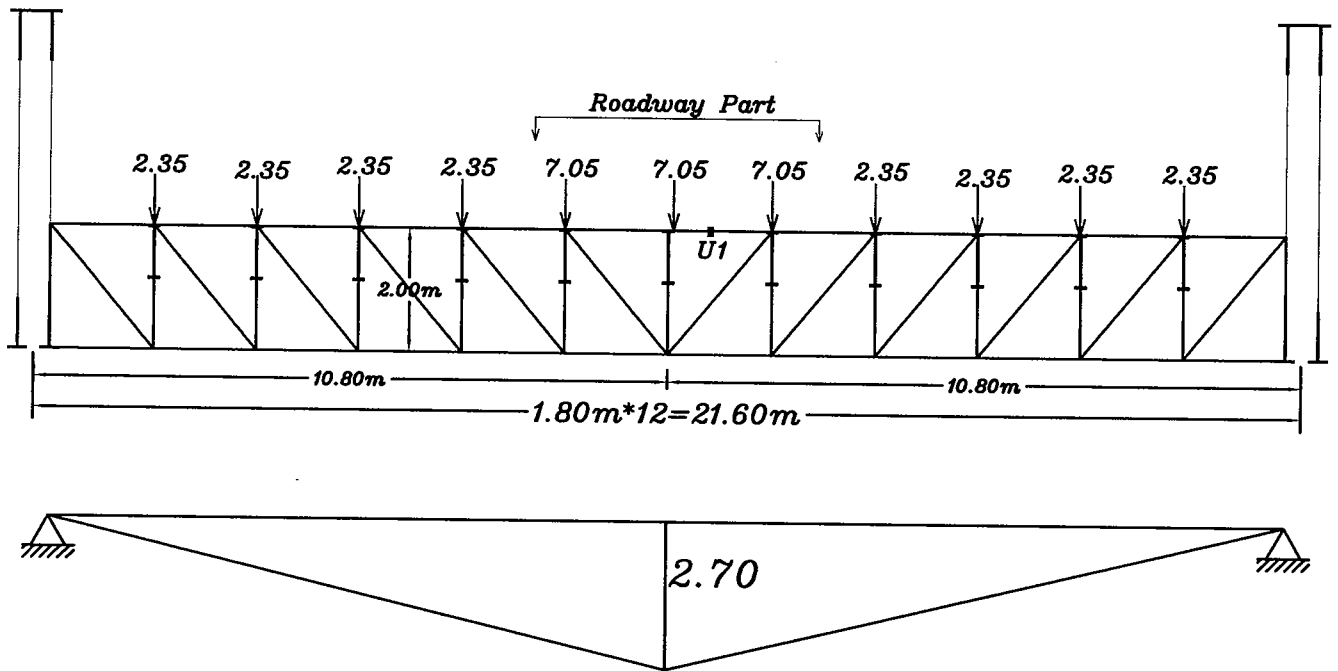
2-ii road way part

$$W_d = (0.21 \times 2.5 + 0.175) \times \overset{\text{Spacing}}{1.80} + \overset{0.W}{0.15}$$

$t_{R.C} \quad \gamma_c \quad F.C$

$$W_d = 1.41t/m'$$

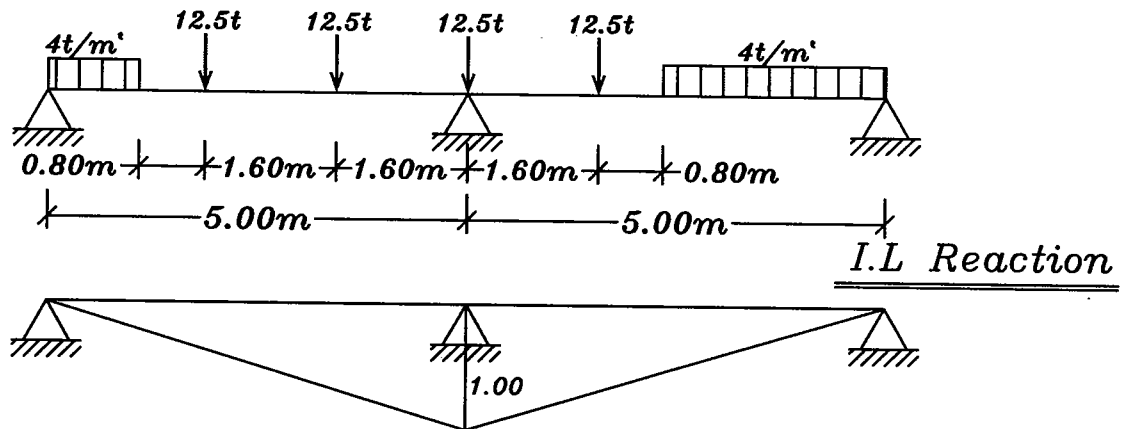
$$R_d = W_d * S = 1.41 * 5.00 = \boxed{7.05t}$$



$$F_{D.L} = 4 \times 2.35 \times 2 \times 1.125 + 2 \times 7.05 \times 2.25 + 7.05 \times 2.70 = 71.91t$$

$$F_{D.L} = 71.91t + 14.58t = \boxed{-86.49t \text{ (comp.)}}$$

upper chord u1
Live Load + impact
Railway Part



$$R_{LL} = 3 \times 12.5 \times 0.680 + 12.5 \times 0.680 + 4 \times 1.0 \times 0.1 + 4 \times 2.6 \times 0.26$$

$$R_{LL} = \boxed{37.10t}$$

$$(1+I) = 0.73 + \frac{2.16}{\sqrt{L_1} - 0.2}$$

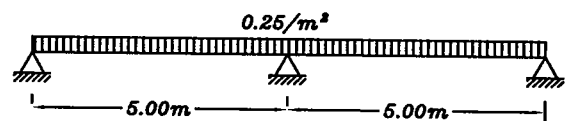
$$L_1 = 2 \times 21.6 = 43.2m$$

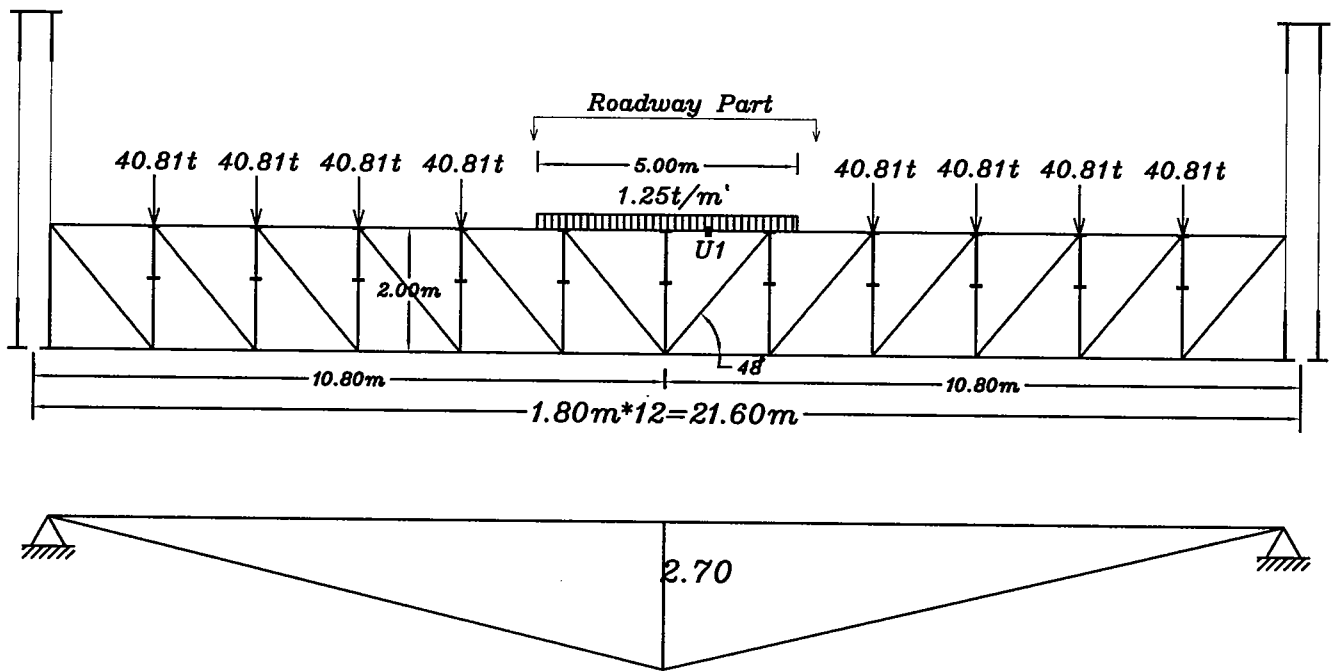
$$(1+I) = 0.73 + \frac{2.16}{\sqrt{43.2} - 0.2} = 1.06 \quad \boxed{< 1.1 \text{ use } 1.1}$$

$$R_{LL+I} = 37.10 \times 1.1 = 40.81t$$

Roadway Part

$$W_{LL} = 0.25 \times 5 = 1.25t/m'$$





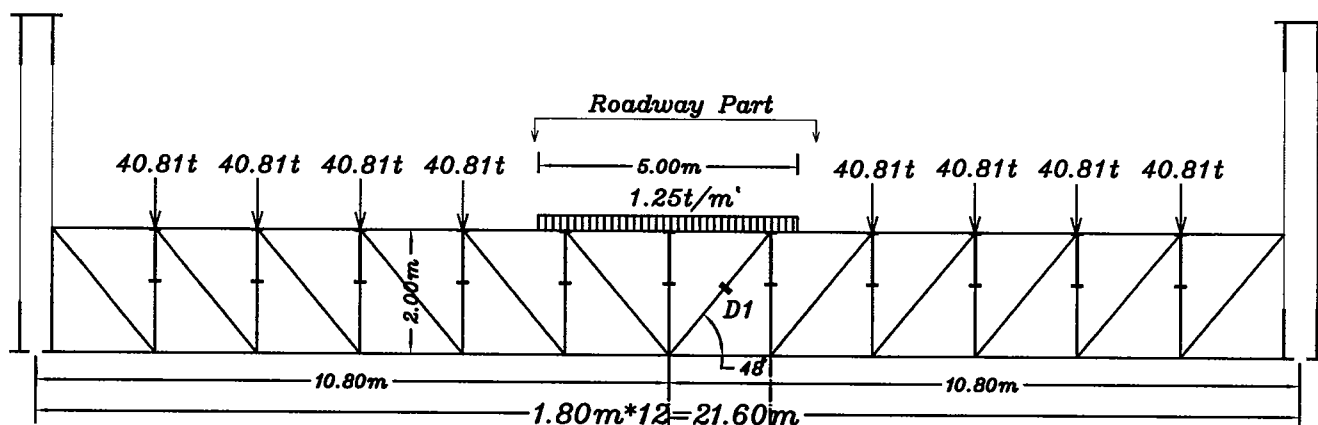
$$F_{L.L+I} = 4 \cdot 40.81 \cdot 2 \cdot 1.125 + 2 \cdot 1.25 \cdot 2.5 \cdot 2.38 = 382.16t$$

Design Values

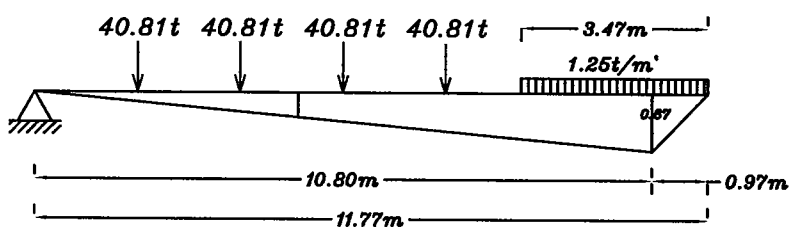
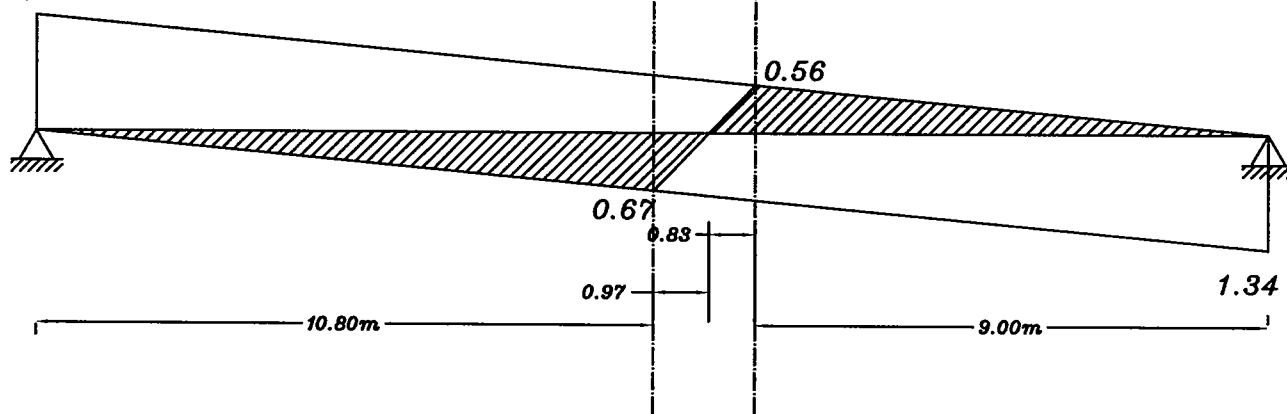
$$F_{max.} = 382.16t + 86.49 = -468.65t \text{ (comp.)}$$

$$F_{min.} = -86.49 \text{ (comp.)}$$

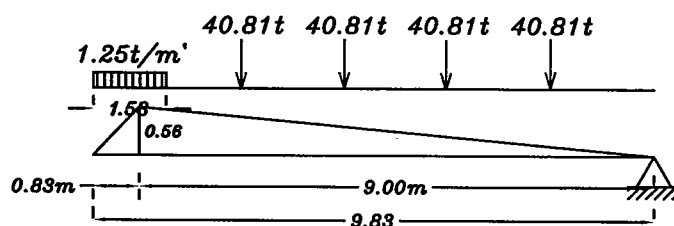
Question 3
Diagonal D2
Live Load + impact



$$1/\sin 48 = 1.34$$



$$F_{L.L+I(tens.)} = 1.25 * 0.97 * 0.335 + 1.25 * 2.5 * 0.59 + 4 * 40.81 * 0.28 = +47.95t$$



$$F_{L.L+I(comp.)} = 1.25 * 0.83 * 0.28 + 1.25 * 0.7 * 0.54 + 4 * 40.81 * 0.28 = -46.4t$$

Question 4

1) assume compression stress to be :

$$F_c = 1.3t/Cm^2 \text{ For St.44}$$

$$h = \frac{\text{Pannel length}}{10 \rightarrow 15} = \frac{180}{10 \rightarrow 15} = 12Cm \text{ to } 18Cm$$

$$\text{use } h = 14Cm$$

$$\text{use } b = 14Cm$$

choose U.P.N 140 channel from tables

$$b' \text{ from tables} = 60 \text{ mm}$$

$$b'' = 14 + 2*6 + 2*10 = 46Cm$$

$$2) F_c = \frac{\text{max. force}}{\text{area}} \quad 1.5 = \frac{468.6}{\text{area}}$$

$$\text{area} = 324.4Cm^2$$

$$A = 2*C + b''*t \quad , \quad 324.4 = 2*20.4 + 46*t \quad , \quad t = 6.16$$

try bigger c channel use U.P.N 240

$$\text{use } h = 24Cm \quad \text{use } b = (0.80 - 1.00)*24 = 18Cm$$

$$\text{use } b' = 8.5Cm \text{ from tables U.P.N 240}$$

$$b'' = 18 + 2*8.5 + 2*10 = 55Cm$$

$$A = 2*C + b''*t \quad , \quad 324.16 = 2*42.3 + 55*t \quad , \quad t = 4.35$$

$$\text{take } t = 4.00Cm$$

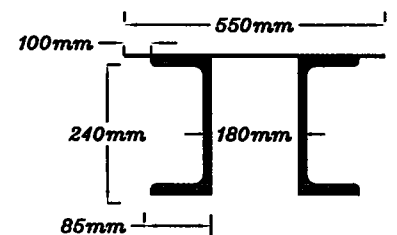
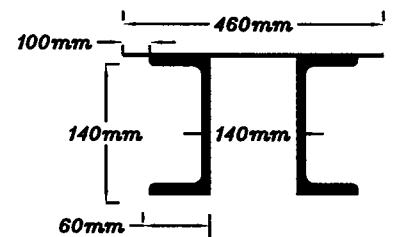
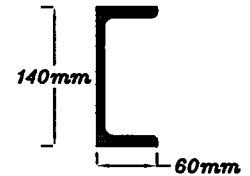
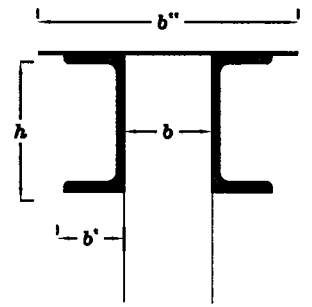
Checks

1-Check Compactness

$$\frac{b}{t} = \frac{18}{4.0} = 4.5 < \frac{64}{\sqrt{2.8}} = 38.2$$

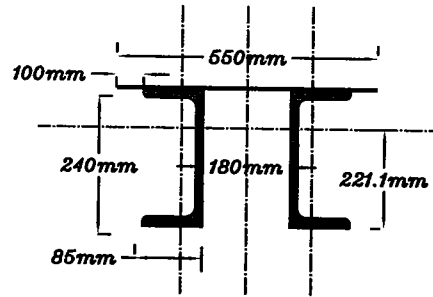
$$\frac{C}{t} = \frac{10}{4.0} = 2.5 < \frac{21}{\sqrt{2.8}} = 2.5$$

channel is non compact sec. from tables



2-Check global buckling

$$y' = \frac{2*42.30*12 + 55*4*(2+24)}{2*42.30 + 55*4} = 22.11 \text{ Cm}$$



$$I_x = 2*(3600 + 42.30*(22.11 - 12)^2) + 55*4(26 - 22.11)^2 = \boxed{19176.20 \text{ Cm}^4}$$

$$I_y = \left[\frac{4.0*55^3}{12} \right] + 2*(248 + 42.30*(11.23)^2) = \boxed{66623.484 \text{ Cm}^4}$$

$$A = 55*4 + 2*42.30 = 304.6 \text{ Cm}^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{19176}{304.6}} = 7.93 \text{ Cm}$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{66623}{304.6}} = 14.79 \text{ Cm}$$

$$\lambda_{in} = \frac{0.85L}{r_x} = \frac{0.85*180}{7.93} = 19.3 < 90$$

$$\lambda_{out} = \frac{L_{out}}{r_y} = \frac{0.85*180}{14.79} = 10.34 < 90$$

3-Check Compressive Stresses

$$\text{actual stresses} = f_{ca} = \frac{468.6}{304} = 1.54 \text{ t/Cm}^2$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * \lambda_{max}^2 \quad \text{For St.44}$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * 19.3^2 = 1.56 \text{ t/Cm}^2$$

Question 5

Design of Upper chord

1) assume compression stress to be :

$$F_c = 1.3t/Cm^2 \text{ For St.44}$$

$$h = \frac{\text{Pannel length}}{10 \rightarrow 15} = \frac{500}{10 \rightarrow 15}$$

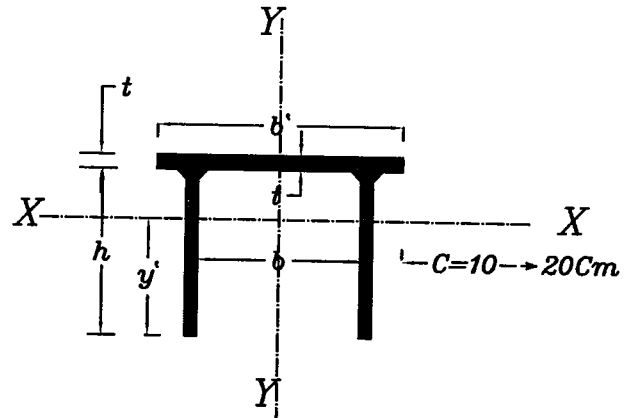
$$h = 50Cm \text{ to } 33.3Cm$$

$$\text{use } h = 45Cm$$

$$\text{use } b = 42Cm \quad \boxed{\text{(given)}}$$

$$b' = b + 2*(10 \rightarrow 20Cm)$$

$$b' = 42 + 2*(10 \rightarrow 20Cm) = 62Cm \text{ to } 82Cm \text{ use } b' = 62Cm$$



$$2) F_c = \frac{\text{max. force}}{\text{area}} \quad 1.3 = \frac{985}{\text{area}}$$

$$\text{area} = 757.7Cm^2$$

$$A = 2ht + b'*t = \dots Cm^2$$

$$757.7 = 2*45*t + 62*t, \quad t = 4.98Cm$$

$$\text{take } t = 5.00Cm$$

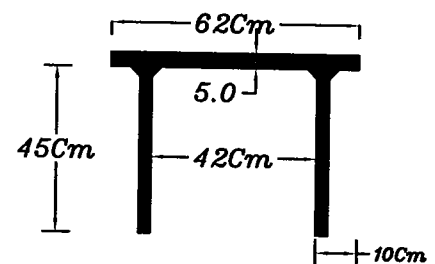
Checks

1-Check Compactness

$$\frac{b}{t} = \frac{42 - 2*1}{5.0} = 8 < \frac{64}{\sqrt{2.8}} = 38.2$$

$$\frac{C}{t} = \frac{10 - 5.0 - 1}{5.0} = 0.8 < \frac{21}{\sqrt{2.8}} = 12.5$$

$$\frac{h}{t} = \frac{45 - 1}{5.0} = 8.80 < \frac{30}{\sqrt{2.8}} = 17.9$$



2-Check global buckling

$$y' = \frac{(2ht*0.5h)+b'*t(h+0.5t)}{2ht+b't} = \dots\dots Cm$$

$$y' = \frac{(45*5.0*22.5*2)+(62*5*47.5)}{(45*5.0*2)+(62*5)} = 32.7Cm$$

$$I_x = 2*\frac{t*h^3}{12} + 2*t*h*(y'-0.5h)^2 + b'*t*(y'-h-0.5t)^2 = \dots\dots Cm^4$$

$$I_x = 2*\frac{5.0*45^3}{12} + 2*5.0*45*(32.7-22.5)^2 + 5.0*62*(47.5-32.7)^2$$

$$I_x = 191304 \quad Cm^4$$

$$I_y = [\frac{5.0*62^3}{12}] + 2*5.0*45*(21+2.5)^2 = 348753 \quad Cm^4$$

$$A = (5.0*2*45) + (62*5.0) = 760 \quad Cm^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{191304}{760}} = 15.87Cm$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{348753}{760}} = 21.42Cm$$

$$\lambda_{in} = \frac{0.85L}{r_x} = \frac{0.85*500}{15.87} = 26.78 < 90$$

$$\lambda_{out} = \frac{L_{out}}{r_y} = \frac{1144.62}{21.42} = 53.43 < 90$$

$$L_{out} = 2.5\sqrt[4]{E*I_y * \alpha * \delta}$$

$$L_{out} = 2.5\sqrt[4]{2100*348753*500*0.12} = 1144.62 \quad Cm$$

3-Check Compressive Stresses

$$\text{actual stresses} = f_{ca} = \frac{985}{760} = 1.29 \quad t/Cm^2$$

$$\text{allwable stresses} = F_c = 1.5 - 7.5*10^{-6} * \lambda_{max}^2 \quad \text{For St.44}$$

$$\text{allwable stresses} = F_c = 1.5 - 7.5*10^{-6} * 53.4^2 = 1.28 \quad t/Cm^2$$

لاحظ انه تم تغيير المعادله التي يتم منها حساب ال F_c وذلك لان تخانة ال Plate

اكبر من ٥ سم (27)

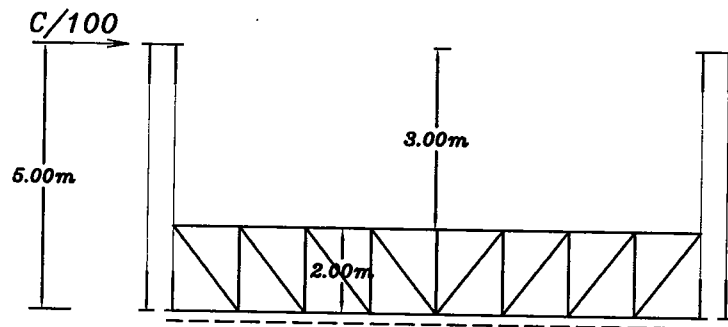
Design of Vertical member

$$F_{v.max} = -F_D * \sin \alpha = 144 * \sin 45^\circ = -102 \text{ (Comp.)}$$

$$F_{v.min} = -F_D * \sin \alpha = 76 * \sin 45^\circ = +54 \text{ (tens.)}$$

Calculation Of moment affecting Vertical member

1-Elastic Force



Where C is the max comp. force in the upper chord

$$M_{add} = C/100 * a = 985/100 * 3.0 = \boxed{29.5 \text{ m.t}}$$

$$a = Hm.g - Hx.g = 3.00 \text{ m}$$

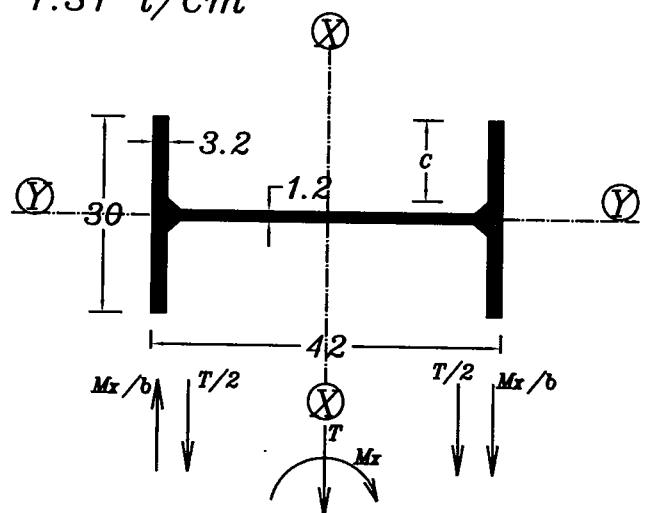
neglect the effect of wind on vertical member

Design as (M+T)

1) assume tension stresses to be :

$$F_{max} = \frac{F_{sr}}{1 - \frac{T_{min}}{T_{max}}} = \frac{2.00}{1 - \frac{54}{-102}} = 1.31 \text{ t/Cm}^2$$

$$F_t = 1.31 \text{ t/Cm}^2$$



$$\text{Force in flange} = \frac{M_x}{b} + \frac{T}{2} = \frac{29.5 \cdot 100}{42} + \frac{54}{2} = 97.23 \text{ t}$$

$$2) F_i = \frac{\text{max. force } T_{D+LL+I}}{\text{area}} \quad 1.31 = \frac{97.2}{\text{area}}$$

$$\text{get } A = 74 \text{ cm}^2$$

$$b = 42 \text{ cm}$$

$$\frac{b}{t_w} = \frac{64}{\sqrt{F_y}} \quad \frac{42}{t_w} = \frac{64}{\sqrt{2.8}} \quad t_w = 1.09 \text{ cm} \quad \text{use } t_w = 1.20 \text{ cm}$$

$$A = b_f \cdot t_f = \dots \text{ cm}^2$$

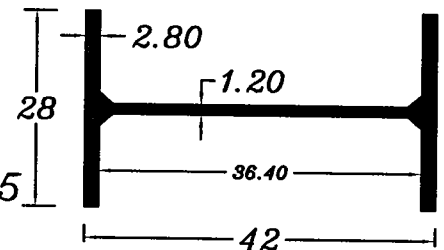
$$b_f \cdot t_f = 74 \text{ cm}^2 \quad \text{take } b_f = 10 \cdot t_f$$

$$t_f = 2.72 \text{ cm} \quad \text{use } t_f = 2.80 \text{ cm} \quad b_f = 74 / 2.80 = 28 \text{ cm}$$

1-Check Compactness

$$\frac{b}{t_w} \leq \frac{64}{\sqrt{F_y}} = \frac{42 - 2 \cdot 2.80 - 2 \cdot 1}{1.20} = 28 < 38$$

$$\frac{C}{t_f} \leq \frac{21}{\sqrt{F_y}} = \frac{14 - 1.2 / 2 - 1}{2.80} = 4.40 < 12.5$$



2-Check global buckling

$$I_x = \frac{t_w \cdot d_w^3}{12} + 2 b_f \cdot t_f \cdot (d_w / 2 + t_f / 2)^2 = \dots \text{ cm}^4$$

$$I_x = \frac{1.2 \cdot 36.4^3}{12} + 2 \cdot 2.8 \cdot 28 \cdot (21 - 1.4)^2 = 65059 \text{ cm}^4$$

$$I_y = 2 \cdot \frac{t_f \cdot b_f^3}{12} = \dots \text{ cm}^4$$

$$I_y = 2 \cdot \frac{2.8 \cdot 28^3}{12} = 10244 \text{ cm}^4$$

$$A = (1.2 \cdot 36.4) + (2 \cdot 28 \cdot 2.8) = 200 \text{ cm}^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{65059}{200}} = 18.1 \text{ Cm}$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{10244}{200}} = 7.15 \text{ Cm}$$

$$\lambda_{in} = \frac{\text{buckling length inplane} = L_{in}}{\text{radius of gyration @ Y axis} = r_y} \nless 90 \quad (\text{RailWay})$$

$$\lambda_{in} = \frac{0.7 \cdot 500}{7.15} = 49 < 90$$

$$\lambda_{out} = \frac{1.20 \cdot 500}{18.10} = 33 < 90$$

3-Check maximum Stresses-case of (M+T)

$$\frac{f_{ta}}{F_t} + \frac{f_{bx}}{F_t} A_1 = \dots\dots\dots \nless 1 \text{ use } \boxed{A_1 = 1.00}$$

$$\text{actual stresses} = f_{ta} = \frac{+54}{200} = 0.27 \text{ t/Cm}^2$$

$$f_t = 1.624 \text{ t/Cm}^2$$

$$f_{bx} = M \cdot y / I_x = [(29.5 \cdot 100 \cdot 42 / 2) / 65059] = 0.95 \text{ t/Cm}^2$$

$$\frac{0.27}{1.624} + \frac{0.95}{1.624} = 0.75 < 1.00 \quad \text{safe}$$

$$\text{fatigue stresses } \frac{54 - (-102)}{200} = 0.78 < 2.00 \text{ t/Cm}^2 \quad \text{safe}$$

check for case of (M+N)

$$\text{actual stresses} = f_{ca} = \frac{102}{200} = 0.51 \text{ t/Cm}^2$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5 \cdot 10^{-5} \cdot \lambda_{max}^2 \quad \text{For St.44}$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5 \cdot 10^{-5} \cdot 49.0^2 = 1.40 \text{ t/Cm}^2$$

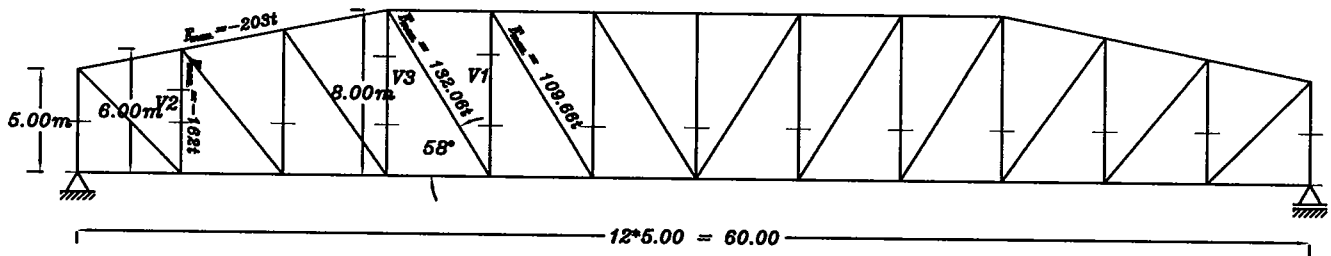
$$\frac{0.51}{1.40} + \frac{0.95}{1.624} = 0.94 < 1.00 \quad \text{safe}$$

Example three

Design of steel Bridges

- Material of Construction is steel 44
- any missing data may be reasonably assumed

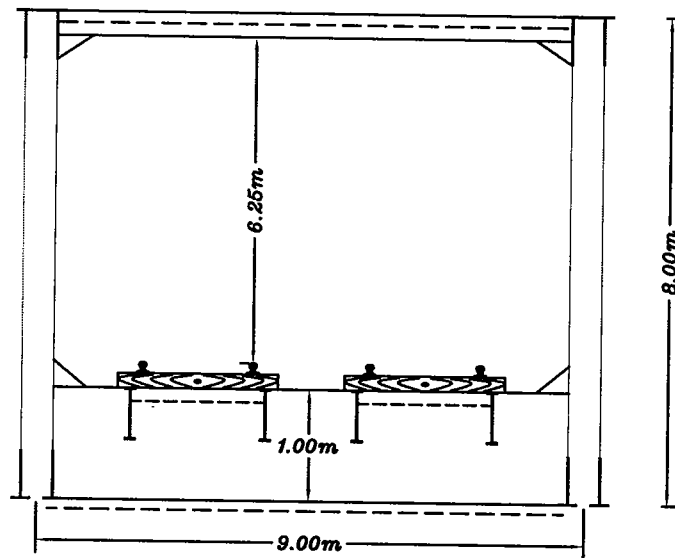
For the Shown Through-Pony bridge Double track the bridge width is 9.00m it is required to :



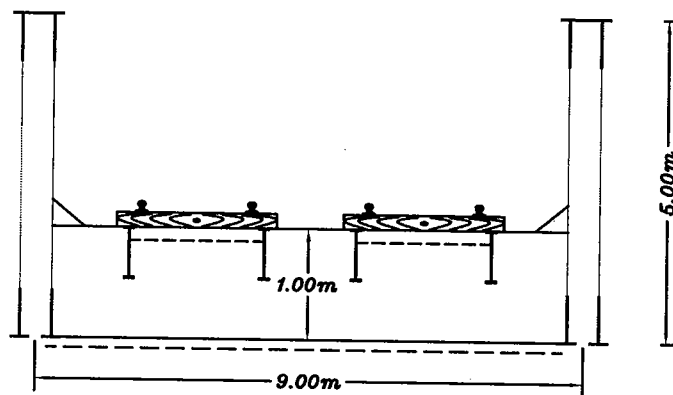
- 1-Complete General Lay out.
- 2-Design the two marked vertical member (V1 & V2) use M27.
- given that the inner distance between the gusset plate = 50Cm
- $F_{sr} = 1.90 \text{ t/Cm}^2$

Example Four

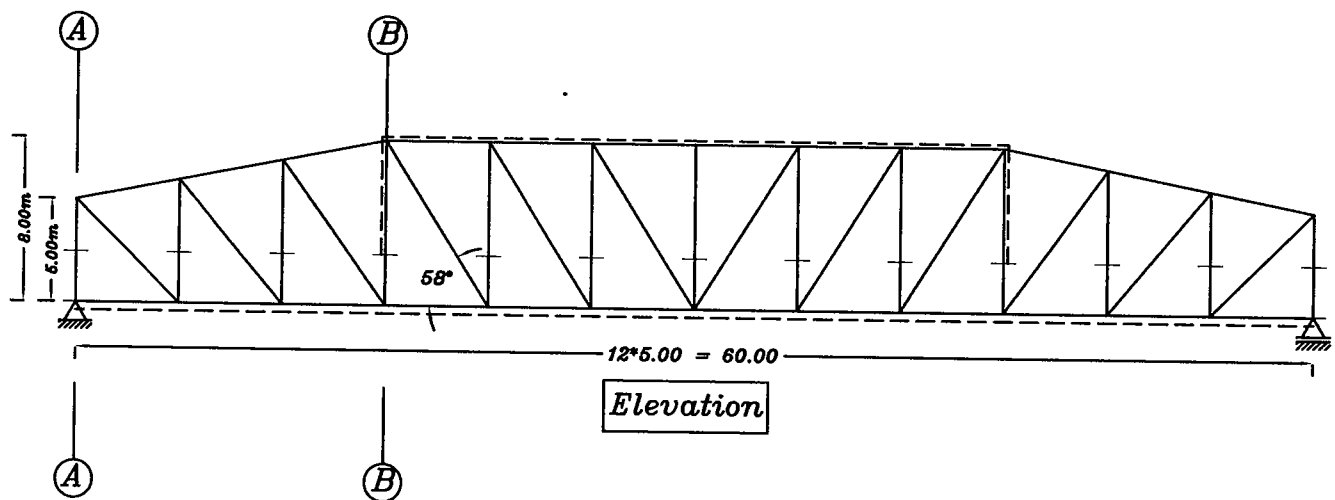
Question 1



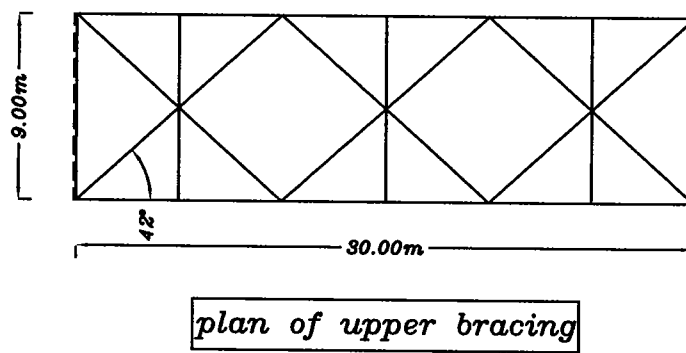
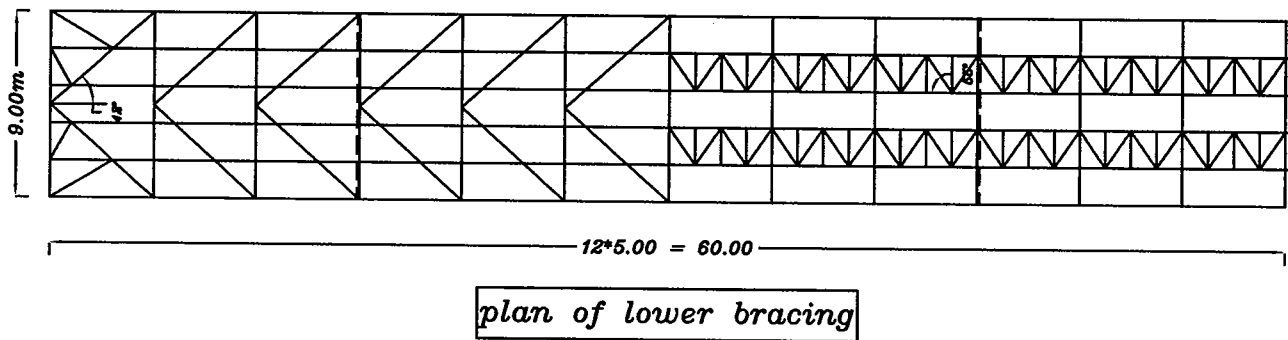
Section B-B



Section A-A



Elevation



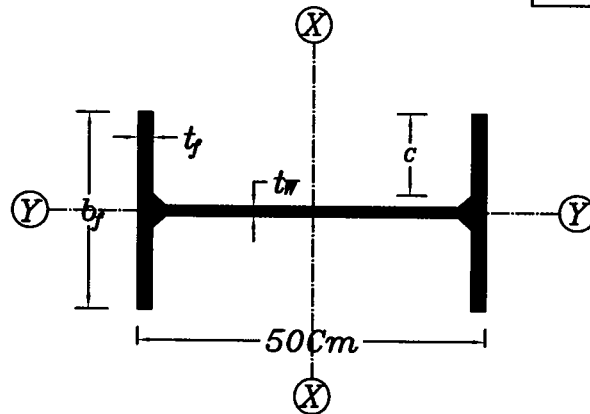
Example Four

Question 2

Design of Vertical member (V1)

this member is vertical in through bridge so it is subjected to axial force only (compression Force)

$$F_{v,max} = -F_D * \sin \alpha = -109.66 * \sin 58^\circ = \boxed{-93 \text{ (Comp.)}}$$



1) assume compression stress to be :

$$F_c = 1.1t/\text{Cm}^2 \text{ to } 1.4t/\text{Cm}^2 \text{ For St.44}$$

$$2) F_c = \frac{\text{max. force}}{\text{area}} \quad 1.1 = \frac{93}{\text{area}} \quad A = 84 \text{ Cm}^2$$

$$b = 50 \text{ Cm} \text{ given}$$

$$\frac{b}{t_w} = \frac{64}{\sqrt{F_y}} \quad \frac{50}{t_w} = \frac{64}{\sqrt{2.8}} \quad t = 1.30 \text{ Cm} \text{ use } t = 1.40 \text{ Cm}$$

$$A = 2 * b_f * t_f + b * t_w = \dots \text{ Cm}^2$$

$$84 = 50 * 1.40 + 2 * b_f * t_f$$

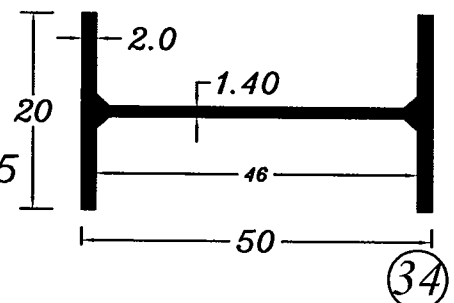
$$b_f * t_f = 14.0 \text{ Cm}^2 \quad \text{take } b_f = \text{min.}, \quad t_f = \text{min.}$$

$$b_{f \text{ min.}} = 6\phi + t_w + 2S \cong 20 \text{ Cm} \quad \text{use } t_f = 2.00 \text{ Cm}$$

1-Check Compactness

$$\frac{b}{t_w} \leq \frac{64}{\sqrt{F_y}} = \frac{50 - 2 * 2.0 - 2 * 1}{1.40} = 31.4 < 38$$

$$\frac{C}{t_f} \leq \frac{21}{\sqrt{F_y}} = \frac{20 - 0.70 - 1}{3.20} = 5.70 < 12.5$$



2-Check global buckling

$$I_x = \frac{t_w * d_w^3}{12} + 2b_f * t_f * (d_w/2 + t_f/2)^2 = \dots \text{Cm}^4$$

$$I_x = \frac{1.4 * 46^3}{12} + 2 * 2.0 * 20 * (25 - 1.0)^2 = 57435 \text{ Cm}^4$$

$$I_y = 2 * \frac{t_f * b_f^3}{12} = \dots \text{Cm}^4$$

$$I_y = 2 * \frac{2.0 * 20^3}{12} = 2666.67 \text{ Cm}^4$$

$$A = (1.4 * 46) + (2 * 20 * 2.0) = 144 \text{ Cm}^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{57435}{144}} = 19.97 \text{ Cm}$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{2666.6}{144}} = 4.30 \text{ Cm}$$

$$\lambda_{in} = \frac{\text{buckling length inplane} = L_{in}}{\text{radius of gyration @ Y axis} = r_y} \nless 90 \text{ (RailWay)}$$

$$\lambda_{in} = \frac{0.7 * 800}{4.30} = 130 > 90 \text{ unsafe use } b_f = 28 \text{ Cm} \quad \boxed{r_y = 6.44}$$
$$\boxed{A = 176.4 \text{ Cm}^2}$$

$$\lambda_{in} = \frac{0.7 * 800}{6.44} = 87 < 90 \text{ safe}$$

$$\lambda_{out} = \frac{0.85 * 800}{19.9} = 34 < 90$$

3-Check Compressive Stresses

$$\text{actual stresses} = f_{ca} = \frac{93}{176} = 0.53 \text{ t/Cm}^2$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5 * 10^{-5} * \lambda_{max}^2 \quad \text{For St.44}$$

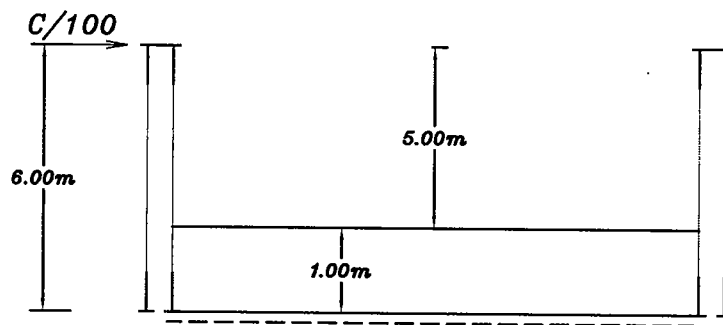
$$\text{allowable stresses} = F_c = 1.6 - 8.5 * 10^{-5} * 87.0^2 = 0.95 \text{ t/Cm}^2$$

Design of Vertical member (V2)

this member is vertical in pony bridge so it is subjected to axial force and moment.

Calculation Of moment affecting Vertical member

1-Elastic Force



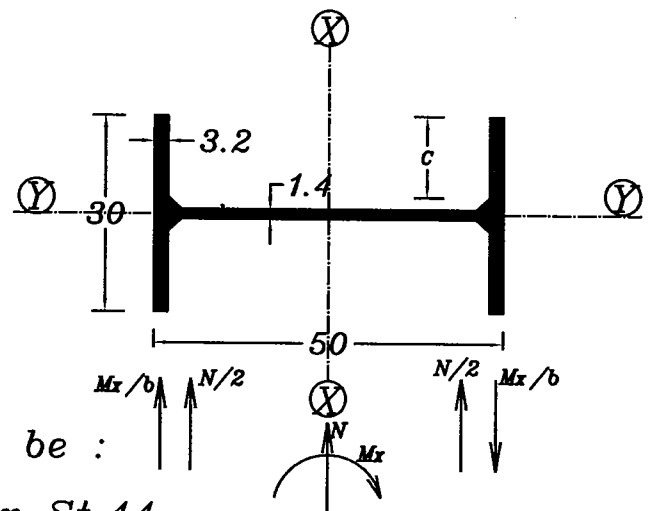
Where C is the max comp. force in the upper chord

$$M_{add} = C/100 * a = 203/100 * 5.0 = \boxed{10.15 m.t}$$

$$a = Hm.g - Hx.g = 5.00 \text{ m}$$

neglect the effect of wind on vertical member

Design as $(M+N)$



1) assume compression stress to be :

$F_c = 1.1t/Cm^2$ to $1.4t/Cm^2$ For St.44

$$\text{Force in flange} = \frac{M_x}{b} + \frac{N}{2} = \frac{10.15 \times 100}{50} + \frac{162}{2} = 101.3 \text{ t}$$

$$2) F_c = \frac{\text{max. force}}{\text{area}} \quad 1.1 = \frac{101.3}{\text{area}} \quad A = 92 \text{ Cm}^2$$

$$b = 50 \text{ Cm} \quad \text{given}$$

$$\frac{b}{t_w} = \frac{64}{\sqrt{F_y}} \quad \frac{50}{t_w} = \frac{64}{\sqrt{2.8}} \quad t = 1.30 \text{ Cm} \quad \text{use } t = 1.40 \text{ Cm}$$

$$A = b_f * t_f = \dots \text{ Cm}^2$$

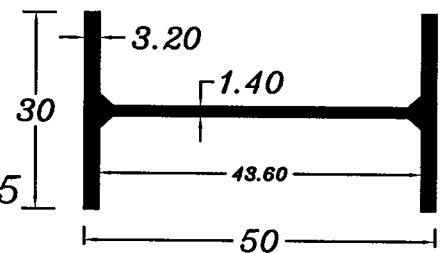
$$b_f * t_f = 92 \text{ Cm}^2 \quad \text{take } b_f = 10 * t_f$$

$$t_f = 3.03 \text{ Cm} \quad \text{use } t_f = 3.20 \text{ Cm} \quad b_f = 92 / 3.20 = 30 \text{ Cm}$$

1-Check Compactness

$$\frac{b}{t_w} \leq \frac{64}{\sqrt{F_y}} = \frac{50 - 2*3.20 - 2*1}{1.40} = 30 < 38$$

$$\frac{C}{t_f} \leq \frac{21}{\sqrt{F_y}} = \frac{15 - 1.4/2 - 1}{3.20} = 4.15 < 12.5$$



2-Check global buckling

$$I_x = \frac{t_w * d_w^3}{12} + 2b_f * t_f * (d_w/2 + t_f/2)^2 = \dots \text{ Cm}^4$$

$$I_x = \frac{1.4 * 43.6^3}{12} + 2 * 3.2 * 30 * (21 - 1.4)^2 = 83428 \text{ Cm}^4$$

$$I_y = 2 * \frac{t_f * b_f^3}{12} = \dots \text{ Cm}^4$$

$$I_y = 2 * \frac{3.2 * 30^3}{12} = 14400 \text{ Cm}^4$$

$$A = (1.4 * 43.6) + (2 * 30 * 3.2) = 253 \text{ Cm}^2$$

$$\text{Calculate } r_x = \sqrt{\frac{I_x}{A}} = \sqrt{\frac{83428}{253}} = 18.1 \text{ Cm}$$

$$\text{Calculate } r_y = \sqrt{\frac{I_y}{A}} = \sqrt{\frac{14400}{253}} = 7.54 \text{ Cm}$$

$$\lambda_{in} = \frac{\text{buckling length inplane} = L_{in}}{\text{radius of gyration @ Y axis} = r_y} \nless 90 \quad (\text{RailWay})$$

$$\lambda_{in} = \frac{0.7*600}{7.54} = 55 < 90$$

$$\lambda_{out} = \frac{1.20*600}{18.10} = 39 < 90$$

check for case of (M+N)

$$\text{actual stresses} = f_{ca} = \frac{162}{253} = 0.64 \text{ t/Cm}^2$$

$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * \lambda_{max}^2 \quad \text{For St.44}$$

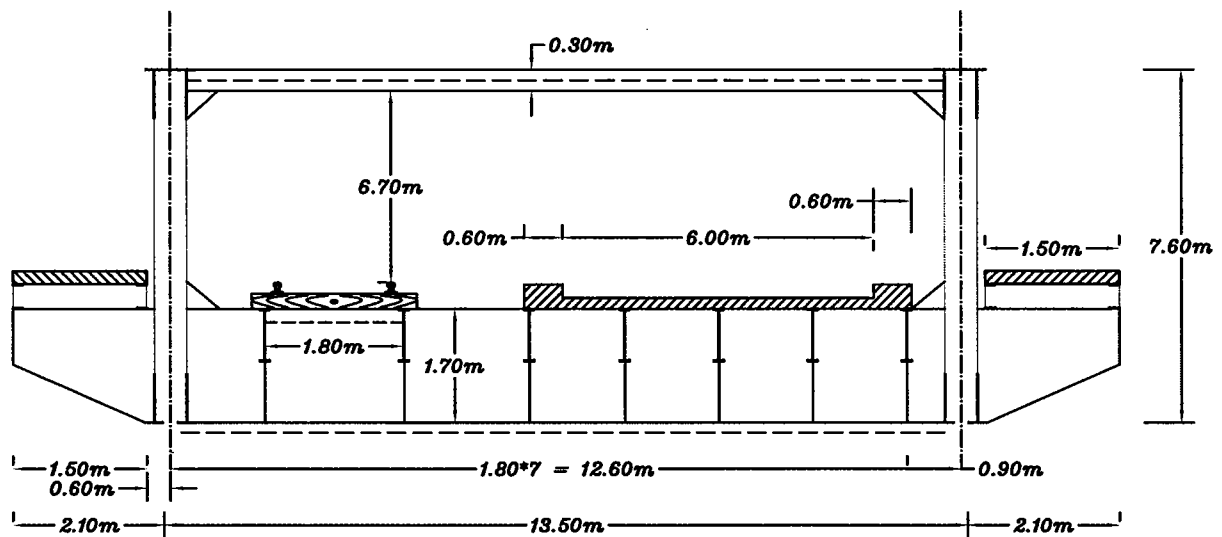
$$\text{allowable stresses} = F_c = 1.6 - 8.5*10^{-5} * 55.0^2 = 1.34 \text{ t/Cm}^2$$

$$f_t = 1.624 \text{ t/Cm}^2$$

$$f_{bx} = M*y/I_x = [(10.1*100*50/2)/83428] = 0.30 \text{ t/Cm}^2$$

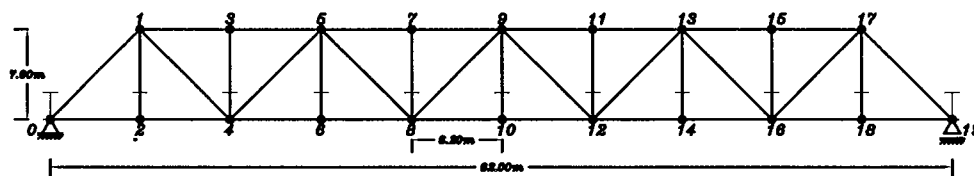
$$\frac{0.64}{1.34} + \frac{0.30}{1.624} = 0.66 < 1.00 \quad \text{safe}$$

Example 4



a rail-road through welded truss bridge of the shown cross section has a span of 62.00m divided into ten equal panels 6.20m each, and depth of 7.60m each as shown in figure it is required to

1-Draw to scale 1:200 a fully Dimensioned layout for the bridge (Elevation, Plans, and cross sections) showing all necessary bracing.



2-Draw the I.L of the diagonal member 4-5 and chord member 6-8.

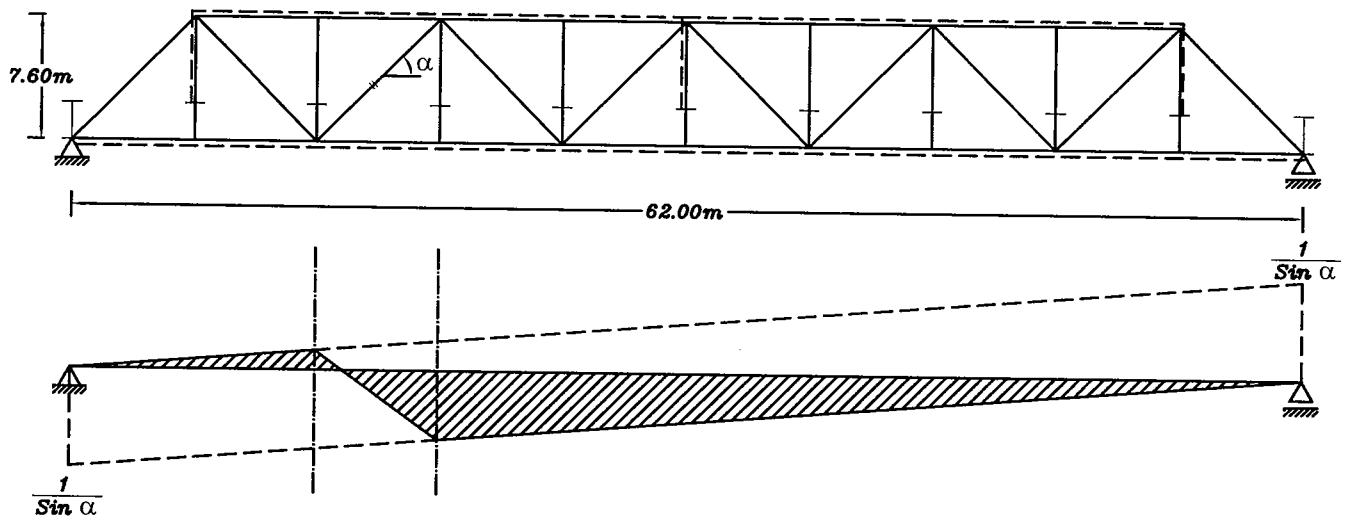
3-calculate force in member 6-8 due to L.L+I only.

4-if all vertical member cancelled state the effect of that.

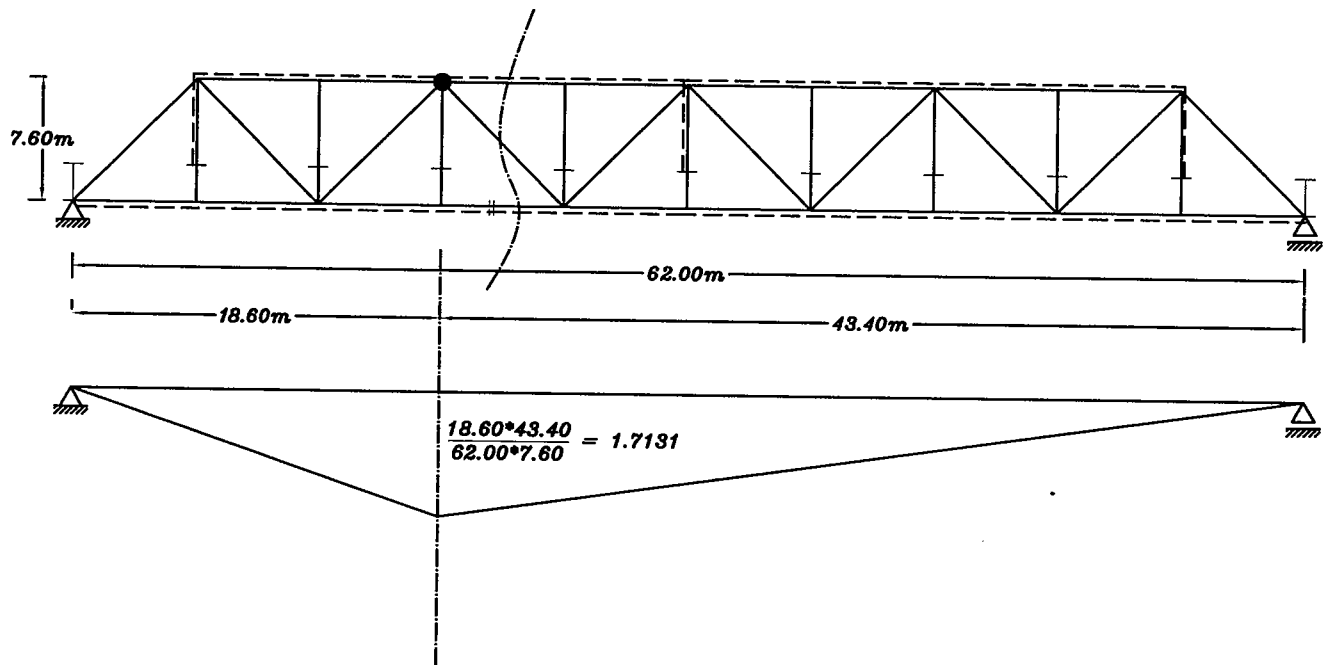
Question 1



Question 2



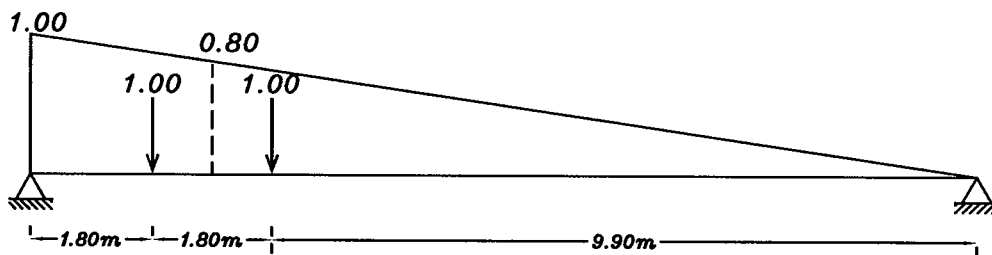
(influence line for member 4-5) (Diagonal member)



(influence line for member 6-8) (Lower Chord member)

Question 3

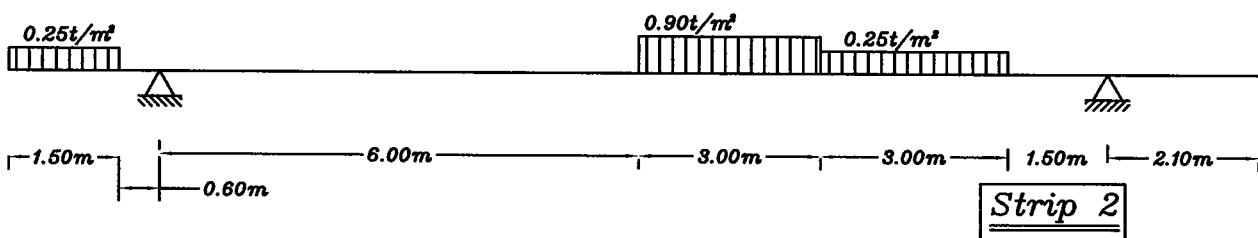
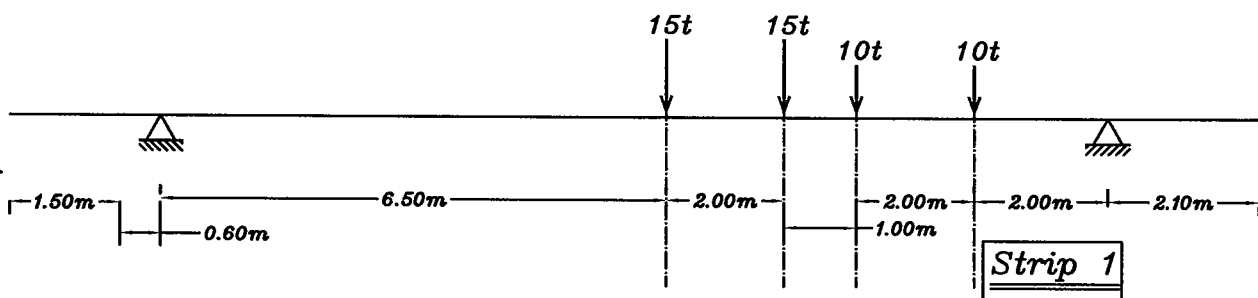
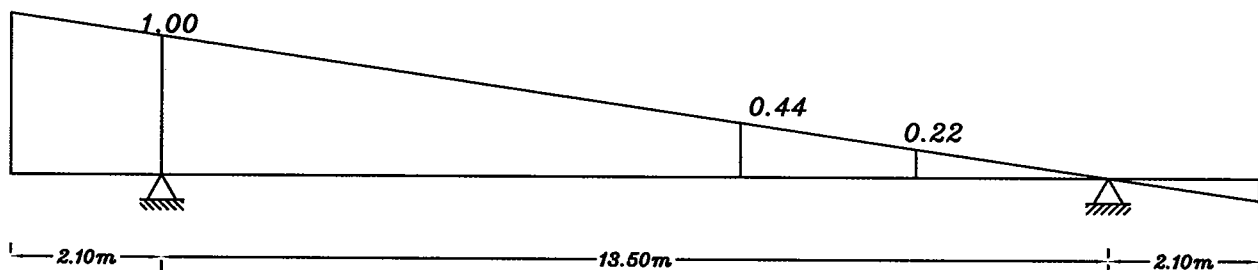
Strip in Railway



$$R = 2 * 1 * 0.8 = \boxed{1.60}$$

$$(1+I) = 0.73 + \frac{2.16}{\sqrt{62.0} - 0.2} = 1.01 \quad \boxed{< 1.1 \text{ use } 1.1}$$

Strip in Roadway



Strip in Roadway

Strip 1

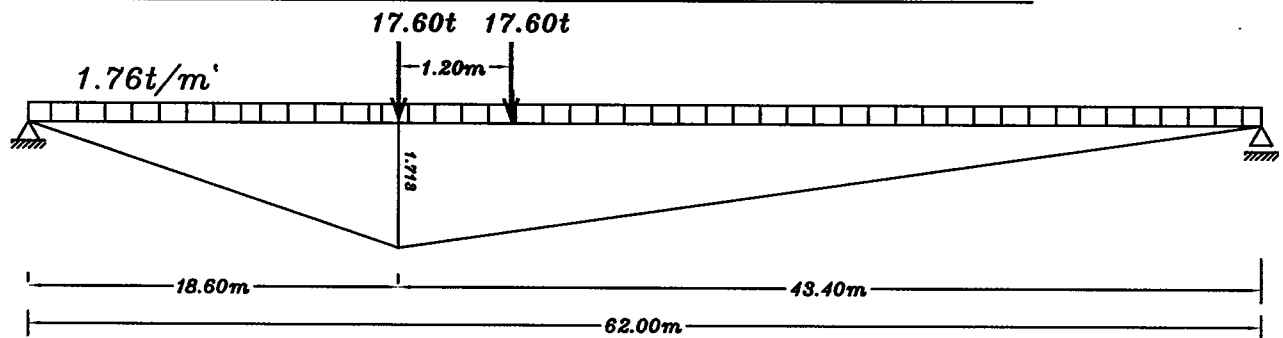
$$R_1 = 2*10*0.22 + 2*15*0.44 = \boxed{17.60t}$$

Strip 2

$$W_2 = 0.25*3*0.22+0.90*3*0.44+0.25*1.5*1.1 = \boxed{1.76t/m'}$$

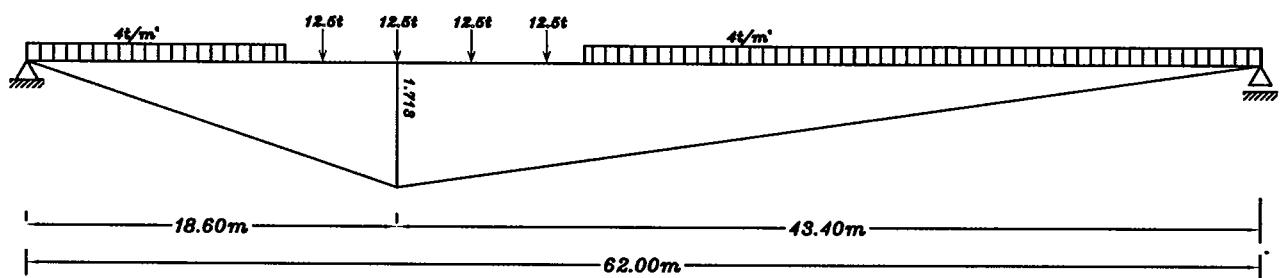
using superposition theory

force in chord member due to roadway only



$$F_{LL+I} = 2*17.60*1.689+0.5*62*1.713*1.76 = 152.914t$$

force in chord member due to railway only

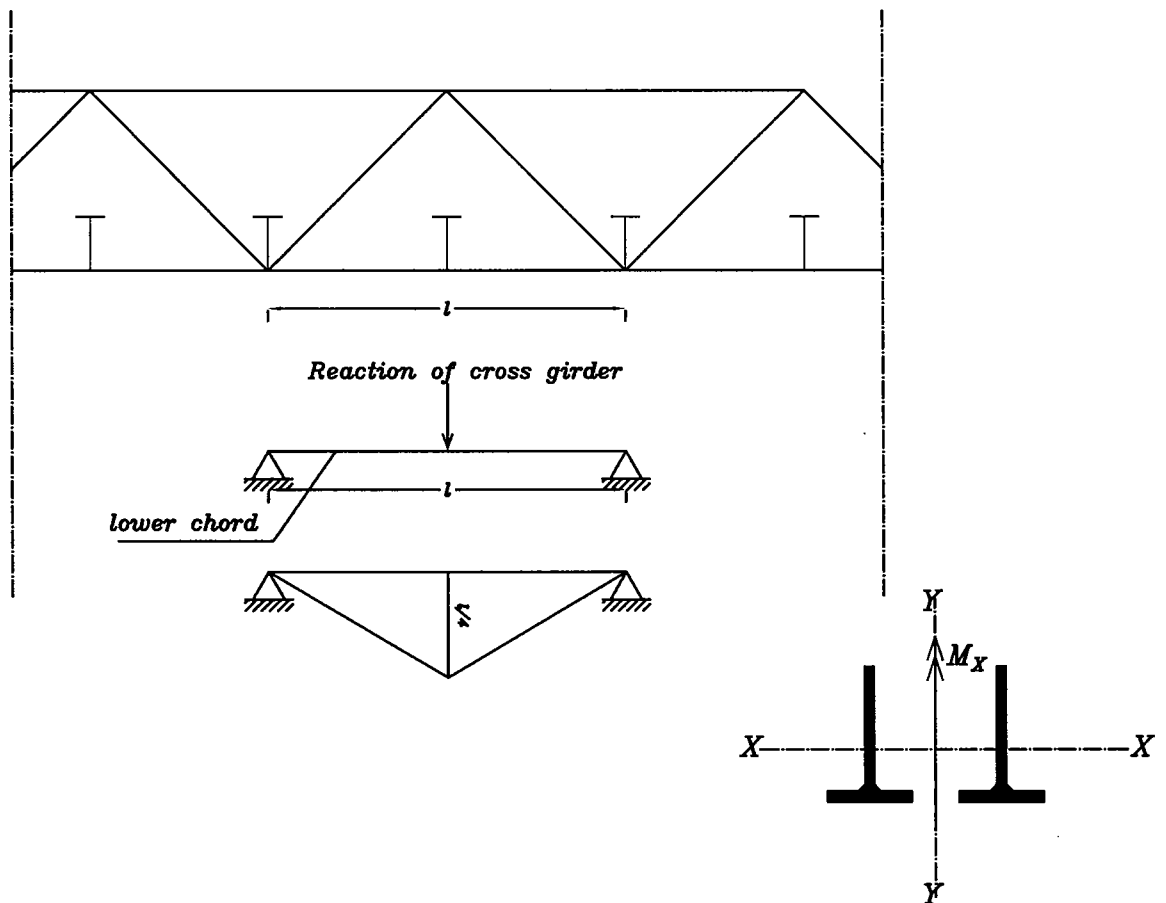


$$F_{LL+I} = 3*12.5*1.645+12.5*1.565+4*(16.2*0.745+39.3*0.775) \\ * 1.10*1.60 = \boxed{442.386t}$$

$$F_{LL+I} = 152.91t + 442.386t = \boxed{595.2t}$$

Road+Rail

Question 4



if all vertical member are cancelled then the the $X.g$ will be a concentrated Load on lower Chord Cause M_x On Lower chord